

Mission Operations

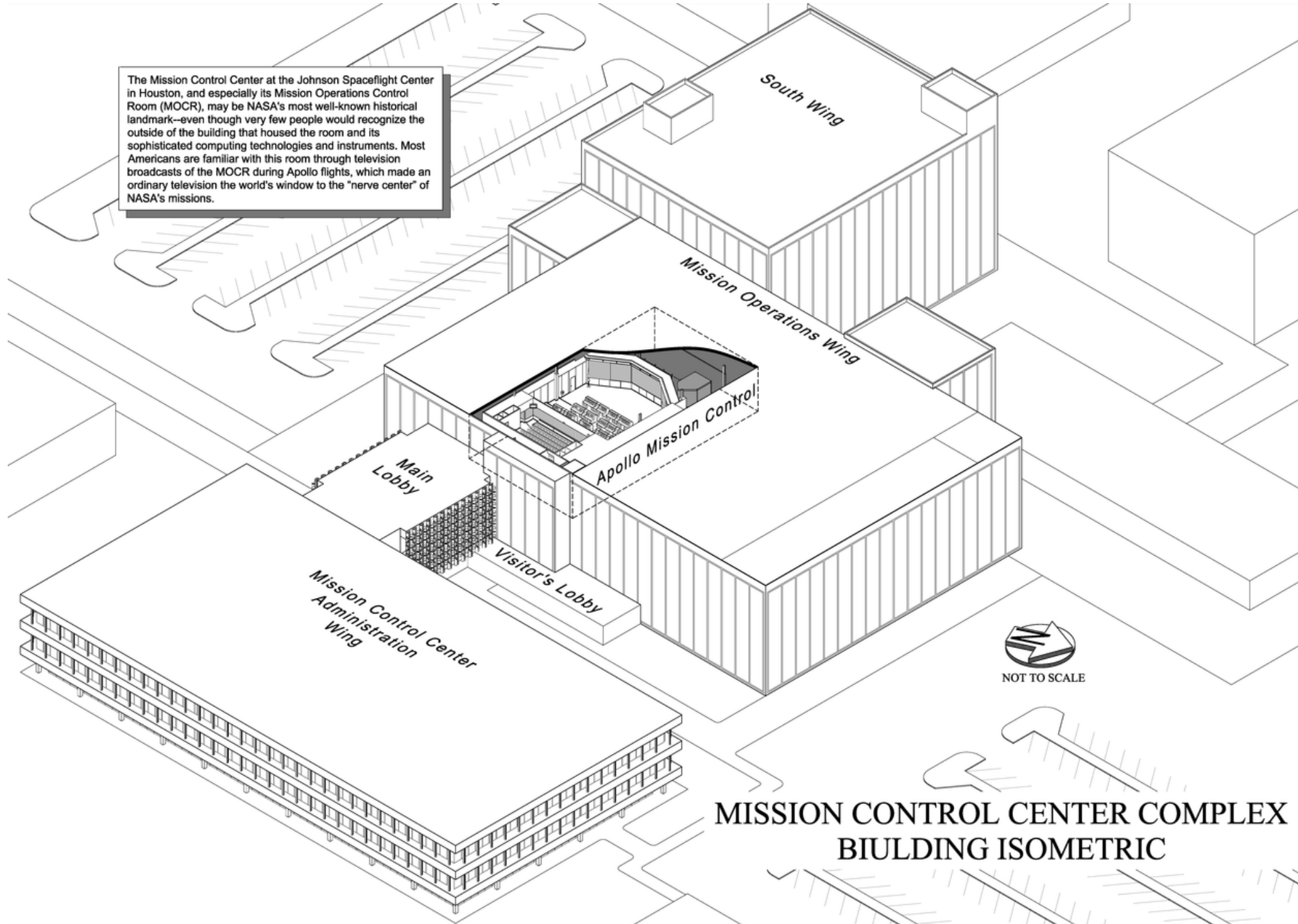
INST 154

Apollo at 50

[The Vital Link](#)



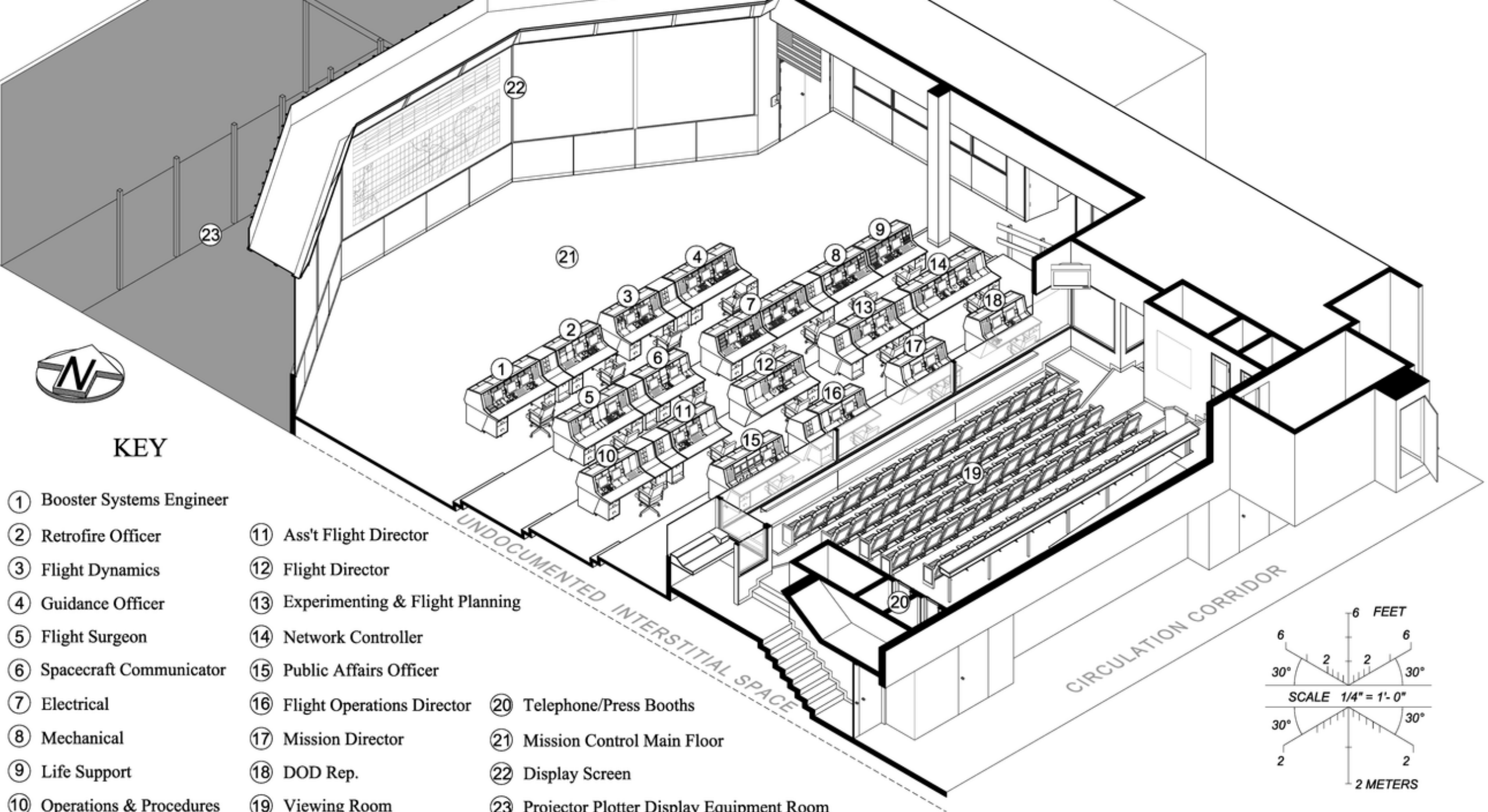
The Mission Control Center at the Johnson Spaceflight Center in Houston, and especially its Mission Operations Control Room (MOCR), may be NASA's most well-known historical landmark—even though very few people would recognize the outside of the building that housed the room and its sophisticated computing technologies and instruments. Most Americans are familiar with this room through television broadcasts of the MOCR during Apollo flights, which made an ordinary television the world's window to the "nerve center" of NASA's missions.



NOT TO SCALE

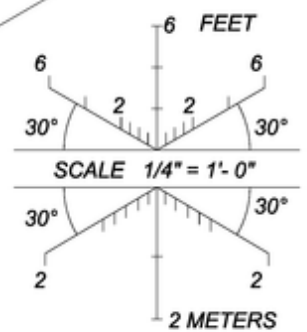
MISSION CONTROL CENTER COMPLEX BIULDING ISOMETRIC

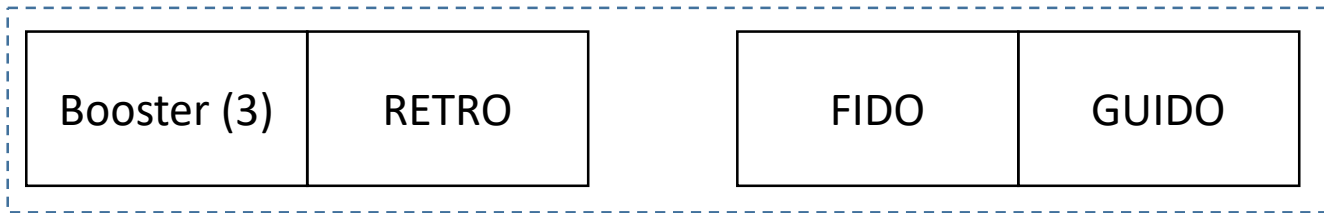




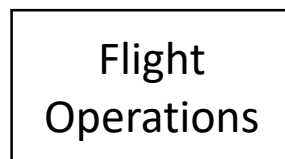
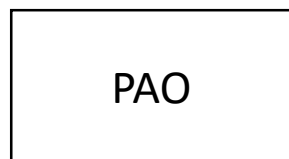
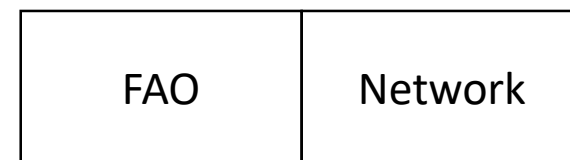
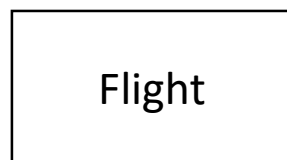
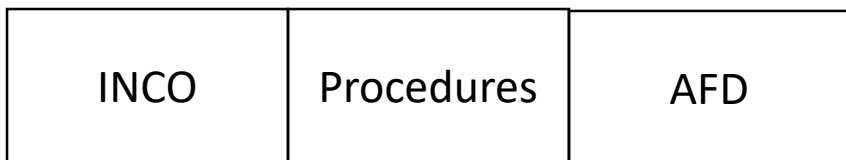
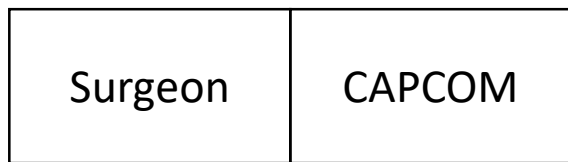
KEY

- | | | |
|-----------------------------------|-----------------------------------|--|
| ① Booster Systems Engineer | ⑪ Ass't Flight Director | ⑳ Telephone/Press Booths |
| ② Retrofire Officer | ⑫ Flight Director | ㉑ Mission Control Main Floor |
| ③ Flight Dynamics | ⑬ Experimenting & Flight Planning | ㉒ Display Screen |
| ④ Guidance Officer | ⑭ Network Controller | ㉓ Projector Plotter Display Equipment Room |
| ⑤ Flight Surgeon | ⑮ Public Affairs Officer | |
| ⑥ Spacecraft Communicator | ⑯ Flight Operations Director | |
| ⑦ Electrical | ⑰ Mission Director | |
| ⑧ Mechanical | | |
| ⑨ Life Support | | |
| ⑩ Operations & Procedures Manager | | |





“The Trench”





NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

MSC-01807
11/1/70

FINAL FLIGHT MISSION RULES

APOLLO 14
(AS-509/110/LM-8)

NOVEMBER 1, 1970

PREPARED BY
FLIGHT CONTROL DIVISION

MANNED SPACECRAFT CENTER
HOUSTON, TEXAS

FOR NASA/DOD INTERNAL USE ONLY
INCLUDING APPROPRIATE CONTRACTORS

INDEXING DATA

DATE	OPR	#	T	PGM	SUBJECT	SIGNATOR	LOC
11-01-70	MSC	MSC-01807	R	AS-509	(206)	MSC	079-37

NASA - Manned Spacecraft Center

MISSION RULES

SECTION 3 MISSION RULE SUMMARY

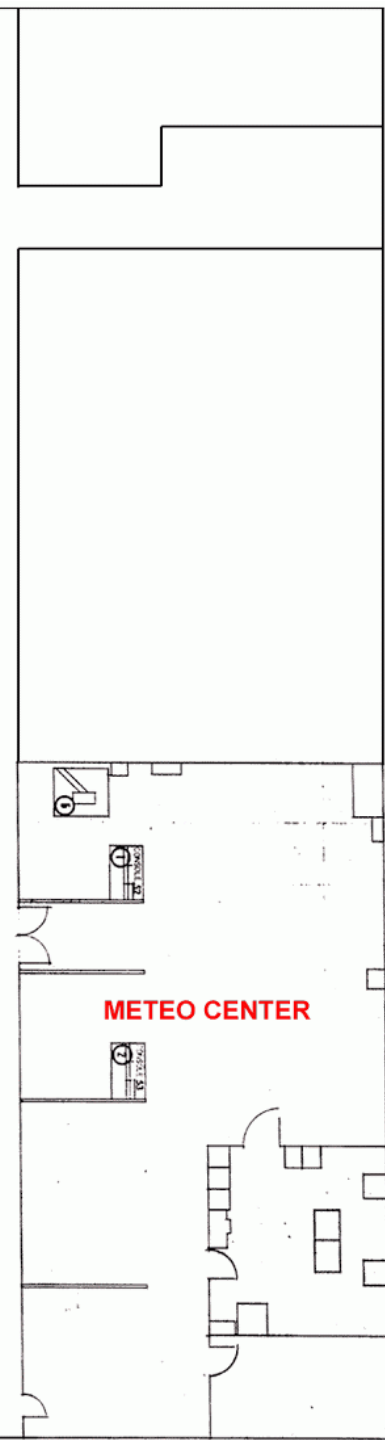
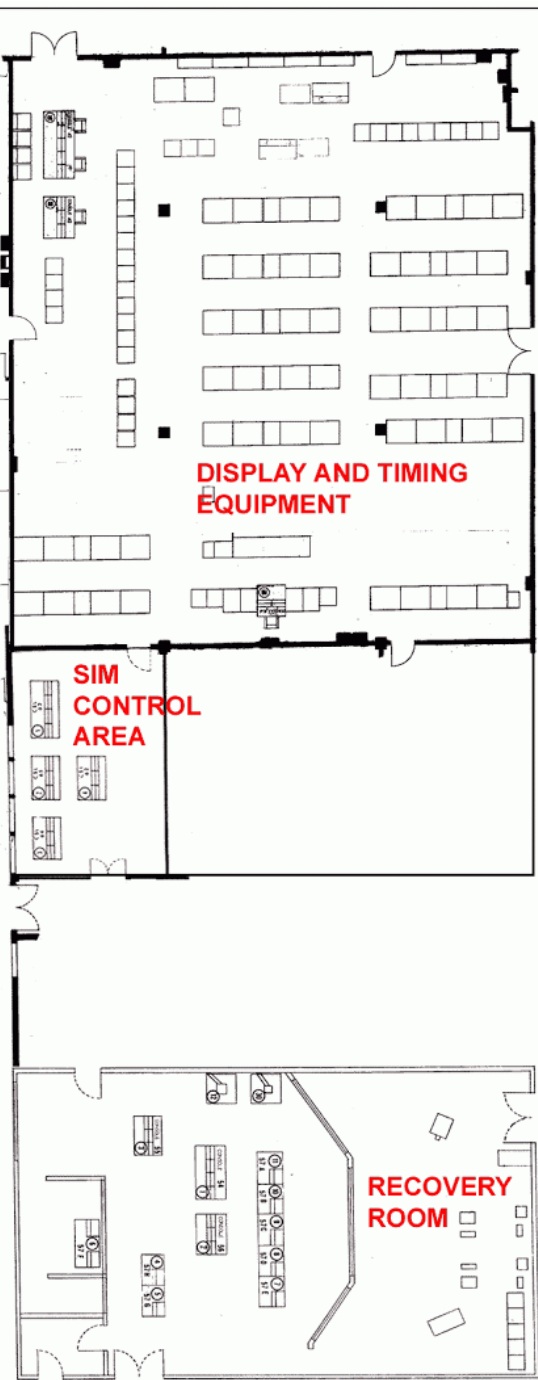
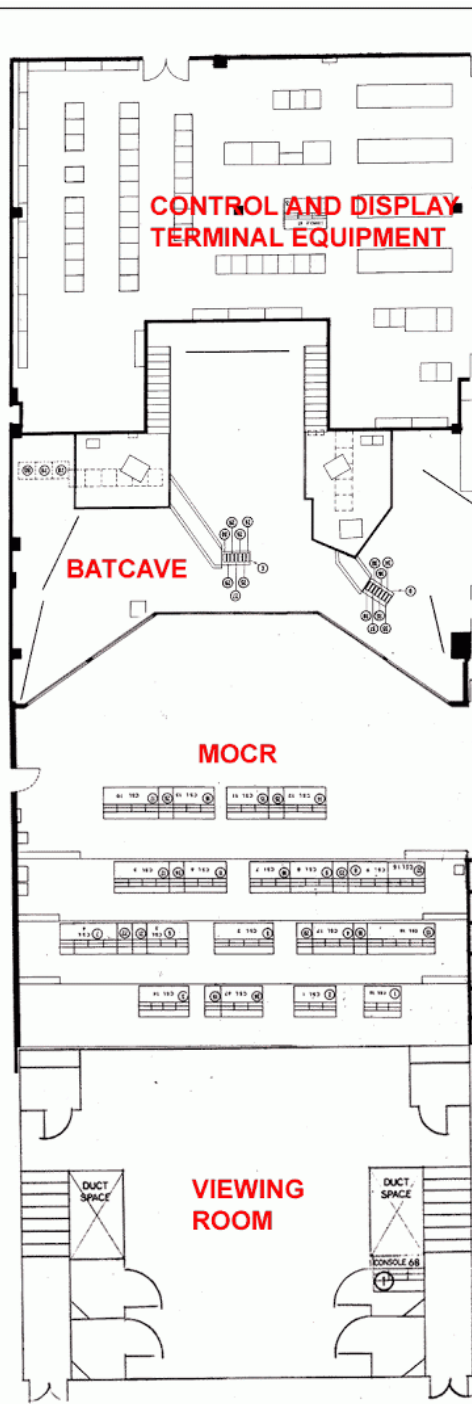
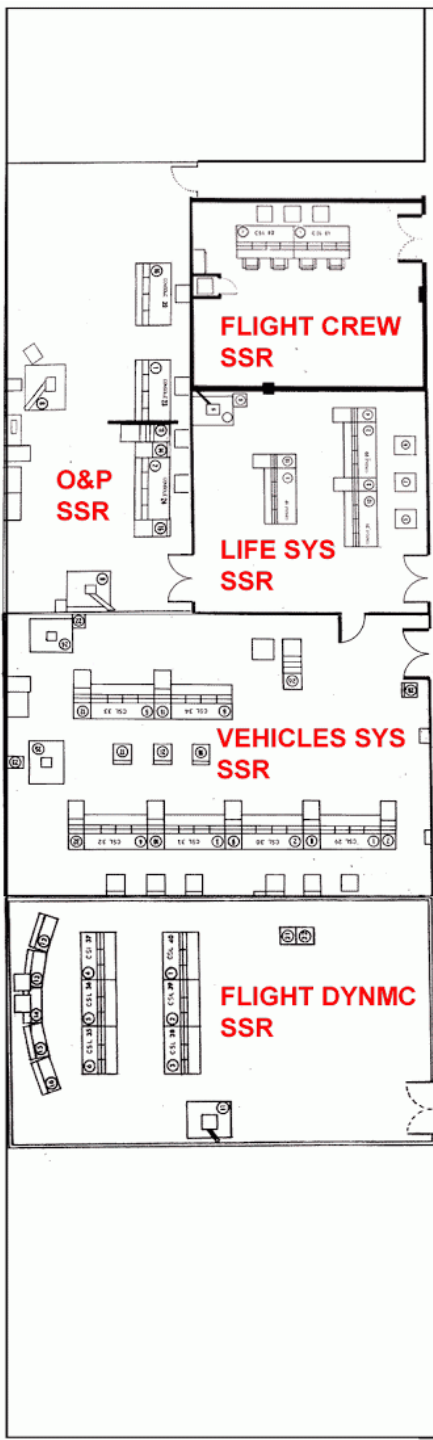
R	ITEM	
		----- ! POWERED DESCENT PHASE ! -----
3-49	PDI IGNITION	THE FOLLOWING ACTION WILL BE TAKEN---
	1.	AUTO ULLAGE GOOD - IF NO AUTO DPS IGN, FLIGHT CREW PERFORM MANUAL DPS IGNITION
	2.	NO AUTO ULLAGE -FLIGHT CREW BACK UP THE ULLAGE MANEUVER -IF NO AUTO DPS IGN FLIGHT CREW WILL NO-GO PDI
3-50	PDI TO LO GATE	POWERED DESCENT WILL BE ABORTED FOR THE FOLLOWING---
	A.	LR DATA IS REQUIRED FOR LANDING - NO LR DATA BY 10 K FT - ABORT.
	1.	LR CONVERGENCE (ALTITUDE ONLY) - DATA NOT BEING ACCEPTED OR CONVERGING FOLLOWING LOCKON FOR 60 SECONDS - ABORT.
	2.	LR DATA ACCEPTED AND CONVERGED CONTINUOUS TO P-64 - CONTINUE MISSION IF LOSS OF LOCK OCCURS IN P-64.
	3.	LR DATA ACCEPTED AND CONVERGED WITH SUBSEQUENT DROPOUT - CONTINUE TO P-64. (A) LANDING RADAR REGAINED IN P-64. (1) DELTA H LESS THAN 1000 FT BETWEEN PGNS AND LR - CONTINUE MISSION. (2) DELTA H GREATER THAN 1000 FT BETWEEN PGNS AND LR - ATTEMPT MANUAL LANDING IN AGS. (B) LR NOT REGAINED IN P-64 - ABORT.
	4.	LATE LR LOCKON WITH DATA BEING INCORPORATED AND CONVERGING - CONTINUE TO P-64. (A) DELTA H LESS THAN 1000 FT BETWEEN PGNS AND LR - CONTINUE MISSION. (B) DELTA H GREATER THAN 1000 FT BETWEEN PGNS AND LR - ATTEMPT MANUAL LANDING IN AGS.
	B.	PGNS ALTITUDE LESS THAN 22,000 FEET AND PNGS NAVIGATION ERRORS, CONFIRMED BY MSFN OR DOPPLER RESIDUALS, THAT CAUSE THE AGS-PGNS RADIAL VELOCITY DIFFERENCE TO EXCEED MINUS 10 FPS, PRIOR TO LANDING RADAR ALTITUDE INCORPORATION AND CONVERGENCE (A MINUS VELOCITY DIFFERENCE INDICATES THAT THE AGS TRAJECTORY IS LOWER THAN THE PNGS TRAJECTORY).
	C.	PGNS NAVIGATION ERRORS, CONFIRMED BY MSFN OR DOPPLER RESIDUALS, THAT RESULT IN THE FOLLOWING AGS-PGNS VELOCITY DIFFERENCES--- DELTA X DOT (DOWNRANGE) GREATER THAN +/- 45 FPS DELTA Y DOT (CROSSRANGE) GREATER THAN +/- 90 FPS DELTA Z DOT (RADIAL) GREATER THAN +/- 35 FPS
	D.	PGNS ALTITUDE LESS THAN 18,000 FEET AND PNGS NAVIGATION ERRORS, CONFIRMED BY DOPPLER BUT NOT BY AGS, CAUSE THE MSFN-PGNS RADIAL VELOCITY DIFFERENCE TO EXCEED MINUS 20 FPS PRIOR TO LANDING RADAR ALTITUDE INCORPORATION AND CONVERGENCE.
	E.	PGNS NAVIGATION ERRORS CONFIRMED BY DOPPLER RESIDUALS BUT NOT BY AGS, THAT RESULT IN THE FOLLOWING MSFN-PGNS VELOCITY DIFFERENCES--- DELTA Y DOT (CROSSRANGE) GREATER THAN +/- 200 FPS. DELTA Z DOT (RADIAL) GREATER THAN +/- 35 FPS.
		NOTE---RULES C AND E ARE INDEPENDENT OF ANY TYPE OF LANDING RADAR UPDATE. FOR RULES B AND C, SWITCHOVER TO AGS WILL BE PERFORMED.
		MISSION REV DATE SECTION GROUP PAGE
		APOLLO 14 FNL 11/1/70 MISSION RULE SUMMARY POWERED DESCENT 3-9

NASA - Manned Spacecraft Center

MISSION RULES

SECTION 5 TRAJECTORY AND GUIDANCE

R	ITEM	
5-89	LR DATA IS REQUIRED FOR LANDING--NO LR DATA BY 10K FT --ABORT.	
	A.	LR CONVERGENCE (ALTITUDE ONLY) - DATA NOT BEING ACCEPTED OR CONVERGING FOLLOWING LOCKON FOR 60 SECONDS - ABORT.
	B.	LR DATA ACCEPTED AND CONVERGED CONTINUOUS TO P-64 - CONTINUE MISSION IF LOSS OF LOCK OCCURS IN P-64.
	C.	LR DATA ACCEPTED AND CONVERGED WITH SUBSEQUENT DROPOUT - CONTINUE TO P-64. 1. LANDING RADAR REGAINED IN P-64. (A) DELTA H LESS THAN 1000FT BETWEEN PGNS AND LR - CONTINUE MISSION. (B) DELTA H GREATER THAN 1000 FT BETWEEN PGNS AND LR - ATTEMPT MANUAL LANDING IN AGS. 2. LR NOT REGAINED AT P-64 - ABORT.
	D.	LATE LR LOCKON WITH DATA BEING INCORPORATED AND CONVERGING - CONTINUE TO P-64. 1. DELTA H LESS THAN 1000 FT BETWEEN PGNS AND LR - CONTINUE MISSION. 2. DELTA H GREATER THAN 1000 FT BETWEEN PGNS AND LR - ATTEMPT MANUAL LANDING IN AGS.
5-90	A.	POWERED DESCENT WILL BE TERMINATED FOR---
	1.	PGNS ALTITUDE LESS THAN 22,000 FEET AND PNGS NAVIGATION ERRORS, CONFIRMED BY MSFN OR DOPPLER RESIDUALS, THAT CAUSE THE AGS-PGNS RADIAL VELOCITY DIFFERENCE TO EXCEED MINUS 10 FPS, PRIOR TO LANDING RADAR ALTITUDE INCORPORATION AND CONVERGENCE (A MINUS VELOCITY DIFFERENCE INDICATES THAT THE AGS TRAJECTORY IS LOWER THAN THE PNGS TRAJECTORY).
	2.	PGNS NAVIGATION ERRORS, CONFIRMED BY MSFN OR DOPPLER RESIDUALS, THAT RESULT IN THE FOLLOWING AGS-PGNS VELOCITY DIFFERENCES--- DELTA X DOT (DOWNRANGE) GREATER THAN +/- 45 FPS DELTA Y DOT (CROSSRANGE) GREATER THAN +/- 90 FPS DELTA Z DOT (RADIAL) GREATER THAN +/- 35 FPS
	3.	PGNS ALTITUDE LESS THAN 18,000 FEET AND PNGS NAVIGATION ERRORS, CONFIRMED BY DOPPLER BUT NOT BY AGS, CAUSE THE MSFN-PGNS RADIAL VELOCITY DIFFERENCE TO EXCEED MINUS 20 FPS PRIOR TO LANDING RADAR ALTITUDE INCORPORATION AND CONVERGENCE.
	4.	PGNS NAVIGATION ERRORS CONFIRMED BY DOPPLER RESIDUALS BUT NOT BY AGS, THAT RESULT IN THE FOLLOWING MSFN-PGNS VELOCITY DIFFERENCES--- DELTA Y DOT (CROSSRANGE) GREATER THAN +/- 200 FPS. DELTA Z DOT (RADIAL) GREATER THAN +/- 35 FPS.
		NOTE---RULES 2 AND 4 ARE INDEPENDENT OF ANY TYPE OF LANDING RADAR UPDATE. FOR RULES 1 AND 2, SWITCHOVER TO AGS WILL BE PERFORMED.
	5.	COMMANDED THRUST INCREASING PRIOR TO THROTTLE DOWN OR P63 TGO=80 SEC.
	6.	RESERVED
	7.	NO THROTTLE RECOVERY BY P63/664 PROGRAM SWITCH PLUS 15 SEC.
	8.	FAILURE TO ACHIEVE FTP BY NOMINAL TIG +31 SEC. (ABORT AT GTC DIVERGENCE).
	9.	FAILURE TO ENTER P64 WHEN TGO EQUALS 60 SECONDS.
	10.	THE FOLLOWING PNGS ALARMS---20105, 00214, 20430, 20607, 21103, 01107, 21204, 21302, 21501, 00402 (CONTINUING).
	B.	POWERED DESCENT MANEUVER WILL BE TERMINATED AND AN ABORT REQUESTED IF THE TIME BIASED DPS ABORT BOUNDARY IS VIOLATED.
5-91		AN ABORT WILL NOT BE REQUESTED FOR A PNGS FAILURE AFTER A PNGS INDICATION THAT THE HIGH GATE TARGETING CONDITIONS HAVE BEEN ACHIEVED.
		MISSION REV DATE SECTION GROUP PAGE
		APOLLO 14 FNL 11/1/70 TRAJECTORY AND GUIDANCE DESCENT 5-12





Program	Number of crewmen	Simulator time, hr (a)	Simulator time per crewman (average), hr	Total training program time, hr	Simulator portion of total training program time, percent
Mercury	7	1 330	190	4 038	33
Gemini	20	6 964	348	17 991	39
Apollo (through mission 15)	32	29 967	936	69 248	43
Total	59	38 261	--	91 277	42

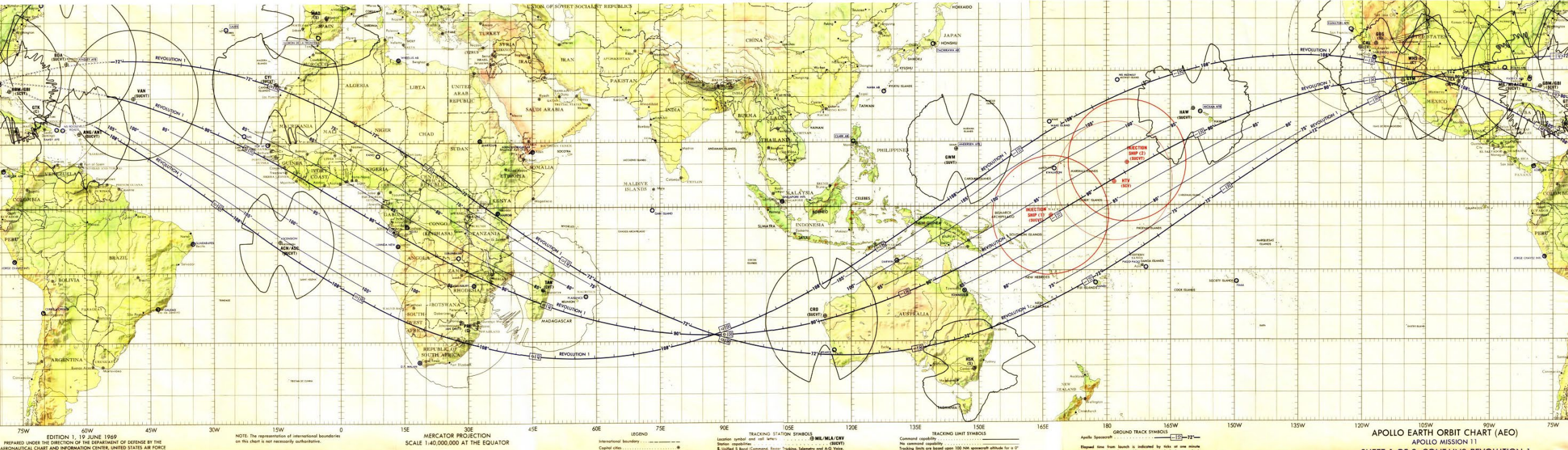
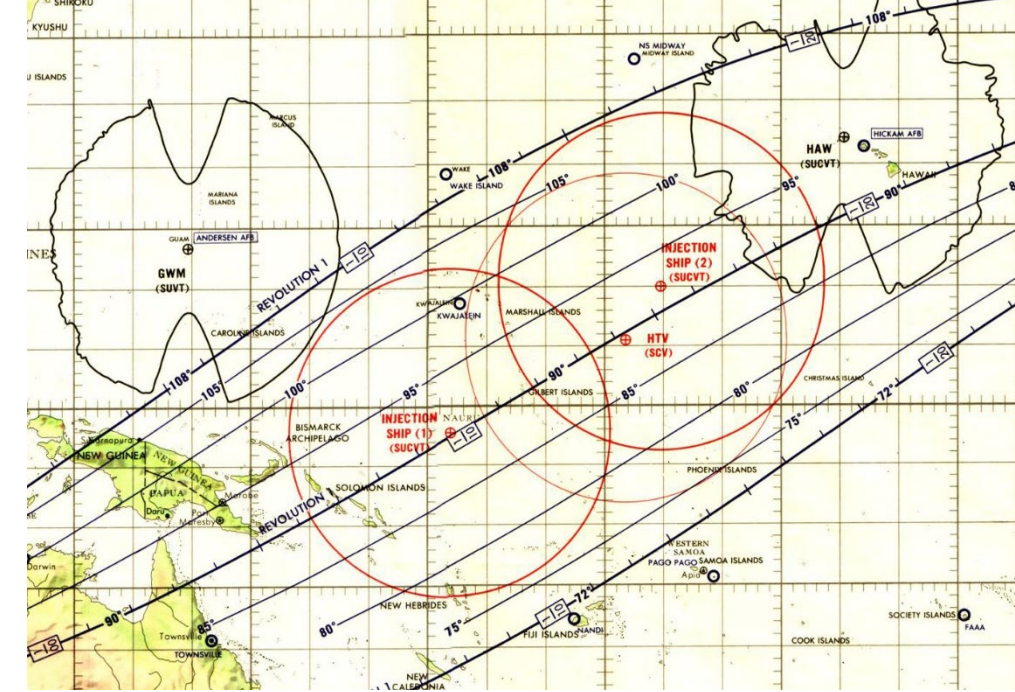
Apollo Mission	Integrated Simulation sessions, days			
	CMS/MCC	LMS/MCC	CMS/LMS/MCC	Total
7	18	0	0	18
8	14	0	0	14
9	10	2	8	20
10	11	0	7	18
11	7	4	7	18
12	10	3	12	25
13	13	5	9	27
14	15	7	13	35
15	19	5	7	31







Manned Space Flight Network



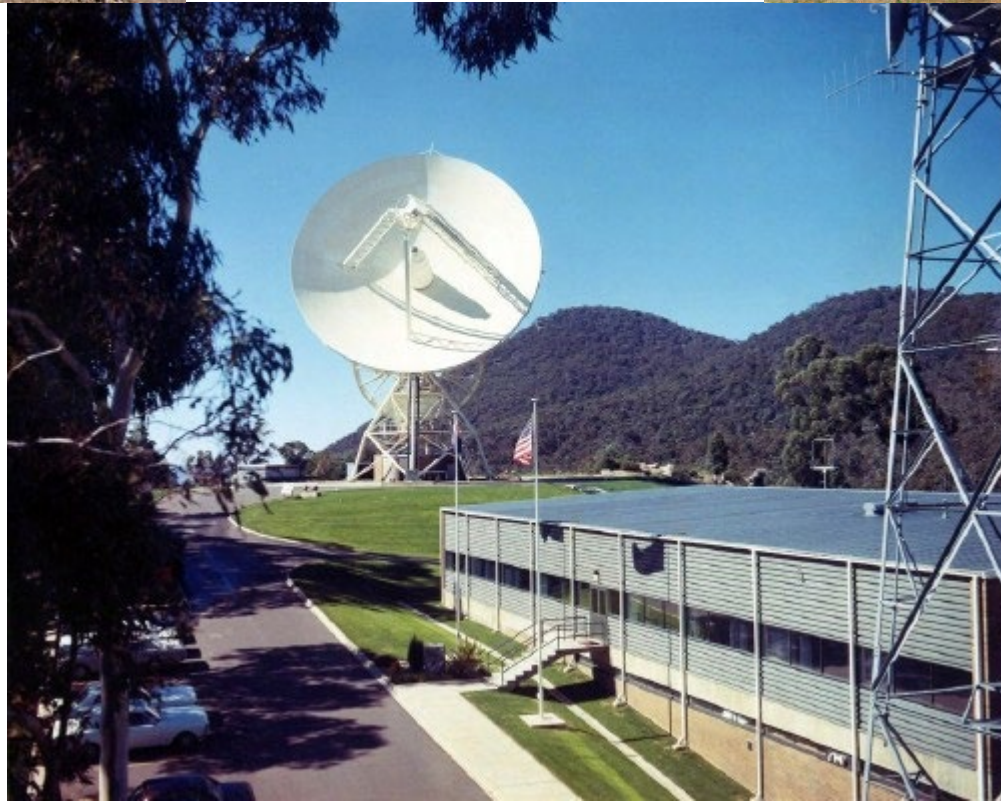
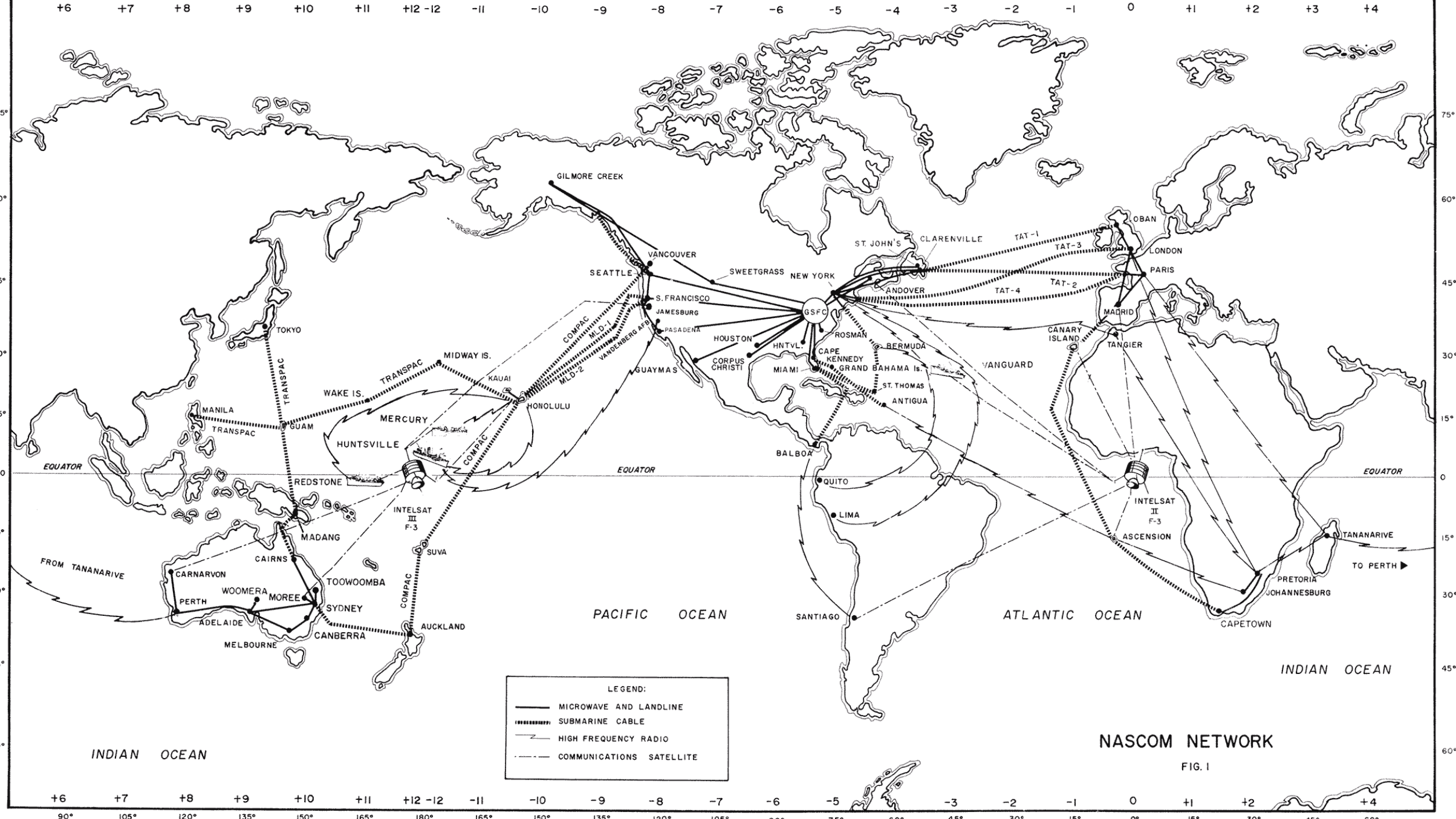


Table 1-1. Network Configuration for AS-512

SYSTEMS STATIONS	TRACKING			USB			TLM			DATA PROCESSING			COMM			OTHER				
	C-band (High-speed)	C-band (Low-speed)	USB	TV to MCC	Voice	TLM	Command	VHF Links	Mag Tape Recording	Decoms	642B TLM	642B CMD	CDP	Acq Computer	High-speed Data	TTY	Voice (SCAMA)	Voice VHF A/G	Range Safety	SPAN
ACN			X		X	X	X	X	X	X	X		X	X	X	X	X			
ANT	X	X																X	X	
ARIA(4)					X	X		X	X							X	X	X		
AOCC																X	X			
BDA	X	X	X		X	X	X	X	X	X	X		X	X	X	X	X	X	X	
CNV	X	X																	X	
CRO	X	X	X		X	X	X	X	X	X	X		X	X	X	X	X	X		X
CYI			X		X	X	X	X	X	X	X		X	X	X	X	X	X		X
GBI	X	X																	X	
GDS			X	X	X	X	X	X	X	X	X		X	X	X	X	X	X		
PIR			X		X	X	X						X							
GTK	X	X																	X	
GWM			X		X	X	X	X	X	X	X		X	X	X	X	X	X		
HAW			X		X	X	X	X	X	X	X		X	X	X	X	X	X		
HSK			X	X	X	X	X	X	X	X	X		X	X	X	X				
NBE			X		X	X	X						X							
MAD			X	X	X	X	X	X	X	X	X		X	X	X	X				
RID			X		X	X	X						X							
MARS				X	X	X														
MIL			X		X	X	X	X	X	X	X		X	X	X	X	X	X		
MLA	X	X																	X	
PARKES				X	X	X														
TEX			X		X	X	X	X	X	X	X		X	X	X	X	X	X		
VAN	X	X	X		X	X	X	X	X	X	X	X	X	X	X	X	X	X		
ETC			X		X	X	X	X	X	X	X					X	X			





LEGEND:

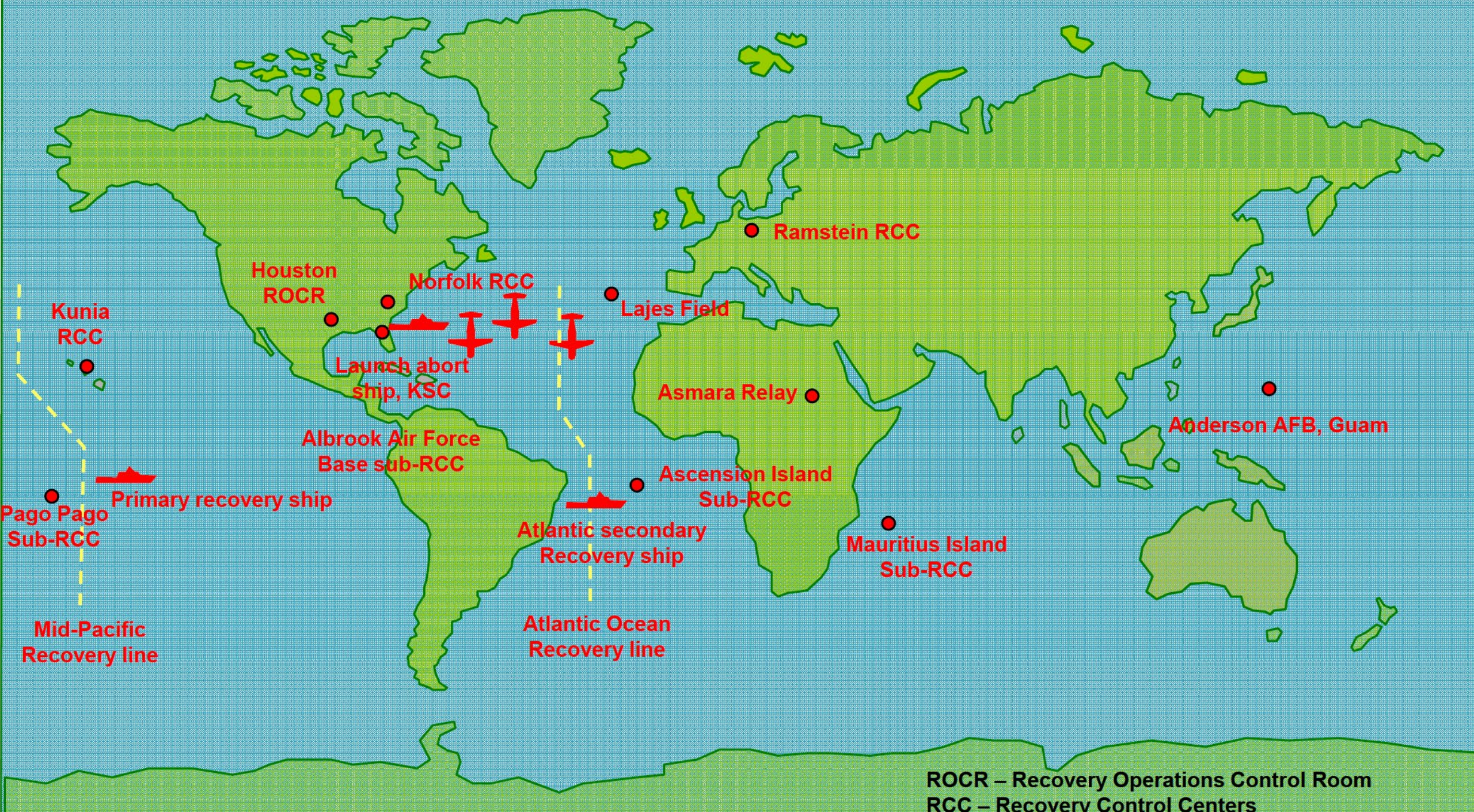
- MICROWAVE AND LANDLINE
- SUBMARINE CABLE
- - - - HIGH FREQUENCY RADIO
- - - - COMMUNICATIONS SATELLITE

NASCOM NETWORK

FIG. 1







ROCR – Recovery Operations Control Room
RCC – Recovery Control Centers

Discussion Groups

- Cox Chapter 19 (“There Will Always Be People Who Want to Work in That Room”)
 - The creation of Mission Control
- Mission Control video
 - Interviews with several flight controllers
- Tsiao Chapter 5 (“The Apollo Years”)
 - Creation of the Manned Space Flight Network

BOOSTER	RETRO
---------	-------

FIDO	GUIDO
------	-------

SURGEON	CAPCOM
---------	--------

EECOM	CNC	TELMU	CONTROL
-------	-----	-------	---------

INCO	O&P	AFD
------	-----	-----

FLIGHT

FAO	NETWORK
-----	---------

PAO

FOD

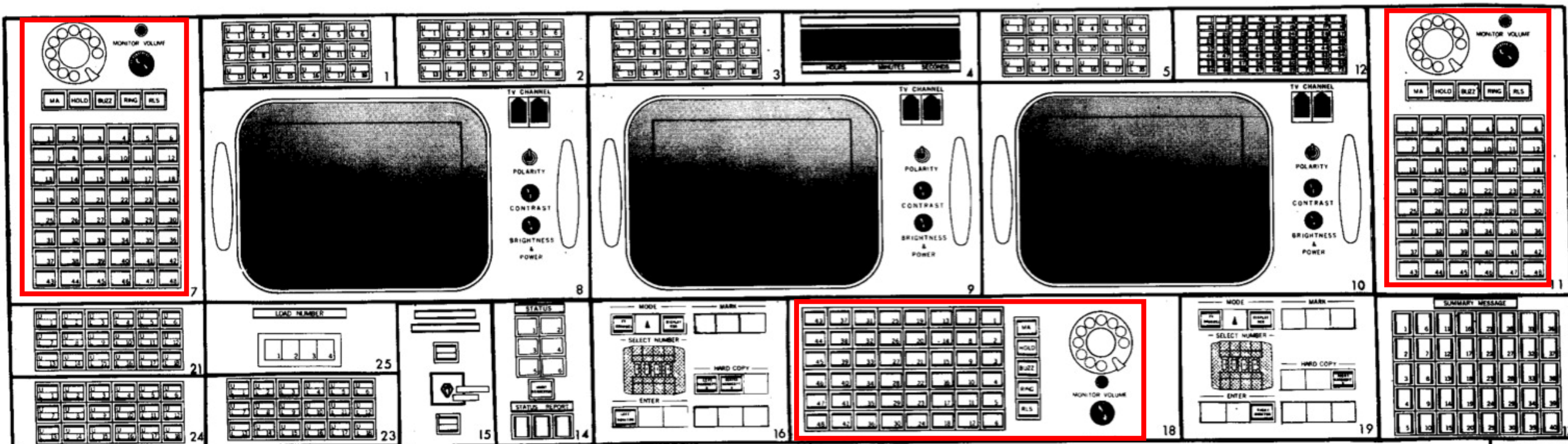
MD

DoD

BOOSTER SYSTEMS
ENGINEER




LOC	DESCRIPTION	TYPE	NOTE	LOC	DESCRIPTION	TYPE
01	EVENT INDICATOR	D9/5B		14	STATUS/STATUS REPORT	D9/1A
02	EVENT INDICATOR	D9/5B		15	TOGGLE SWITCH/INDICATOR	D9/9A
03	EVENT INDICATOR	D9/5B		16	MANUAL SELECT KEYBOARD	A6B/5
04	7 DIGIT CLOCK	D8/3		17	SUMMARY MSG ENABLE KEYBOARD	A19/A
05	EVENT INDICATOR	D9/5B		18	VOICE COMM POSITION-3018	H48MFD
07	VOICE COMM POSITION-3016	V48MFD		19	MANUAL SELECT KEYBOARD	A6B/6
08	TV MONITOR 14" PRECISION	C2/1		21	SWITCH MODULE	D9/40F
09	TV MONITOR 14" PRECISION	C2/1		23	SWITCH MODULE	D9/40E
10	TV MONITOR 14" PRECISION	C2/1		24	SWITCH MODULE	D9/40F
11	VOICE COMM POSITION-3017	V48MFD		25	LOAD NUMBER INDICATOR	D9/41B
12	EVENT INDICATOR (72)	D9/28				





https://apolloinrealtime.org/11



The First Landing on the Moon
Apollo 11
Real-Time Mission Experience
Sat Jul 19 1969 | 11:41:15 PM
Mission Elapsed Time: 086:09:15

Countdown | Earth On the Way to the Moon | In Lunar Orbit | On the Surface | Luna Docked w/ Returning to Earth | Re-entry

In Lunar Orbit | Begin lunar orbit 6/75 | 086:09:15 | Crew Sleep Period Start

In Lunar Orbit 5/75 | Music: Frank Sinatra - It's Nice To Go Trav'ling | Music: Bettye Swann - Angel Of The Morning

Mission Status

Mission Day: 4/9
Mission Phase:
In lunar orbit
Crew Status:
In Command Module, docked with the Lunar Module
Command Module:
In lunar orbit. Orbit: 6/31
Distance from Earth:
207,559 nautical miles (384,399.2 km) average

Mission Control Channels

PHOTOGRAPHY | MISSION CONTROL AUDIO | ASTROMATERIAL SAMPLES

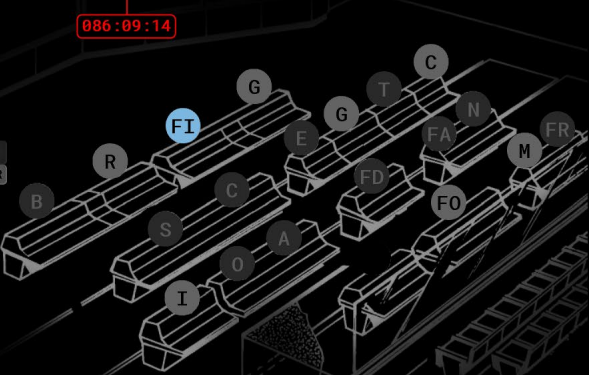
FOD
MSN DIR
FLIGHT
FLIGHT-L
FLIGHT-R
CAPCOM
CAPCOM-R
BOOSTER
BOOSTER-C
BOOSTER-R
RETRO
FIDO
GUIDO
GUIDO-R
SURGEON
SURGEON-R
EECOM
GNC
TELCOM
CONTROL
INCO

086:09:15:07

TRANSCRIPT | MISSION MILESTONES | COMMENTARY

086:08:47 Aldrin Oops! ...
086:12:16 (Music - Bettye Swann - Angel Of The Morning)
086:13:39 Collins What time is it, Neil, 88 hours, something like that?
086:13:42 Armstrong 86:13.
086:15:54 Armstrong Doesn't it look like some of these crater walls had scallops inside like a design in a fan - like feathers.
086:16:05 Aldrin Seashells.
086:16:06 Armstrong Yes.
086:16:07 Aldrin Like seashells - very pretty, very symmetrical.
086:17:37 Armstrong Take along one of those craters.
086:17:57 Aldrin I took overlapping pictures of all that ... as well.
086:18:02 Collins I want to take the - we're going to have to carry a lot

086:09:14



FIDO: Flight Dynamics Officer - Responsible for the flight path of the space vehicle, both atmospheric and orbital. During lunar missions the FIDO was also responsible for the lunar trajectory. The FIDO monitored vehicle performance during the powered flight phase and assessed abort modes, calculated orbital maneuvers and resulting trajectories, and monitored vehicle flight profile and energy levels during re-entry.