

Mission Operations

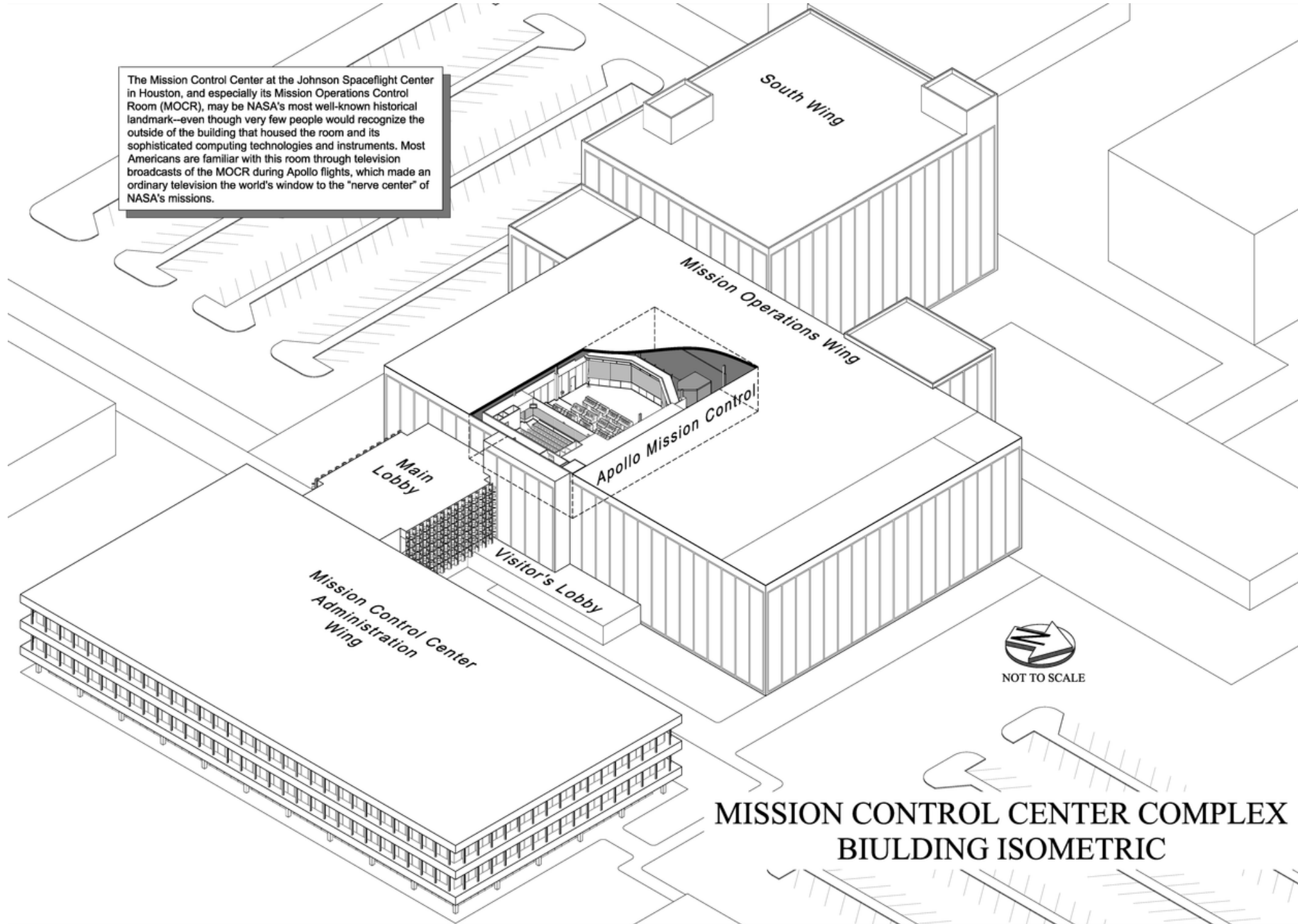
HONR 269i

To the Moon and Back: The Apollo Program

[The Vital Link](#)



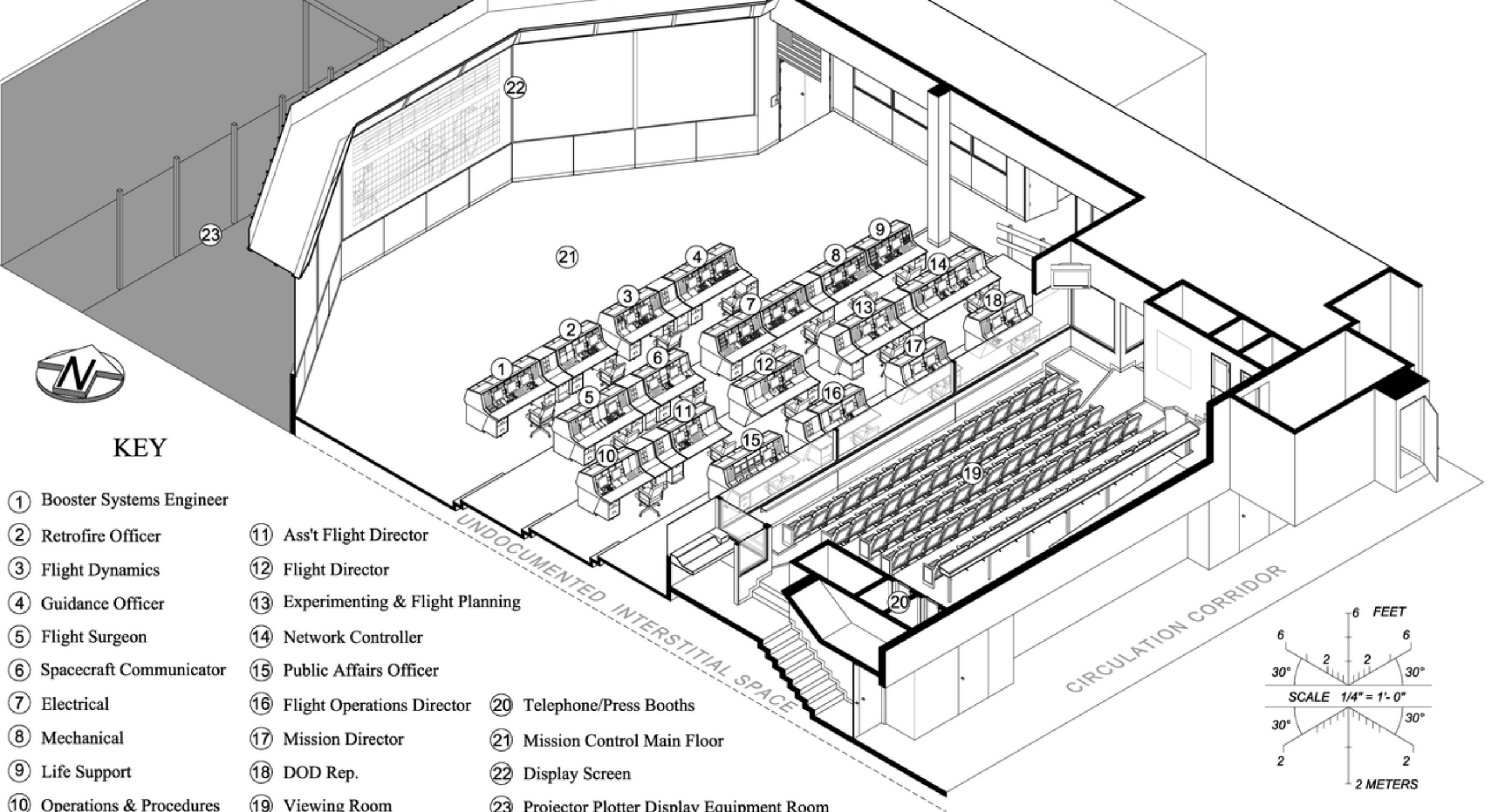
The Mission Control Center at the Johnson Spaceflight Center in Houston, and especially its Mission Operations Control Room (MOCR), may be NASA's most well-known historical landmark—even though very few people would recognize the outside of the building that housed the room and its sophisticated computing technologies and instruments. Most Americans are familiar with this room through television broadcasts of the MOCR during Apollo flights, which made an ordinary television the world's window to the "nerve center" of NASA's missions.



NOT TO SCALE

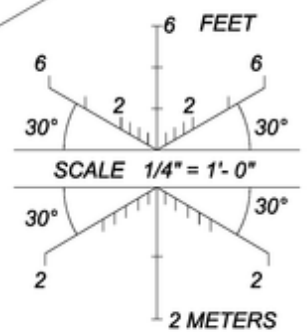
MISSION CONTROL CENTER COMPLEX
BIULDING ISOMETRIC

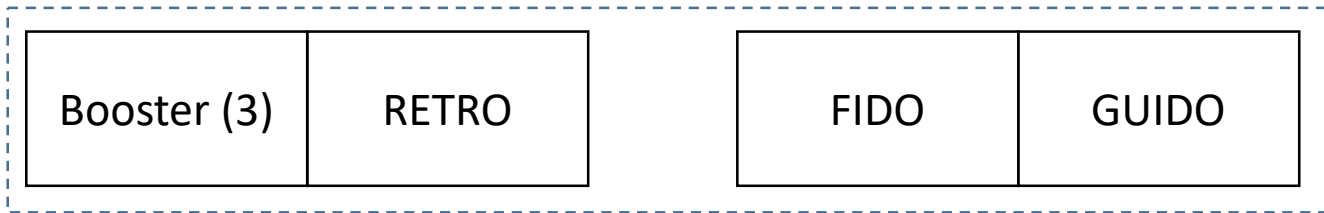




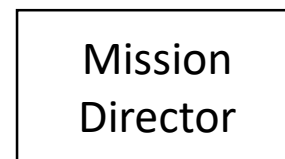
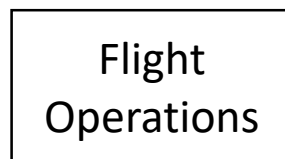
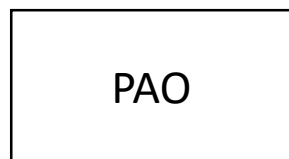
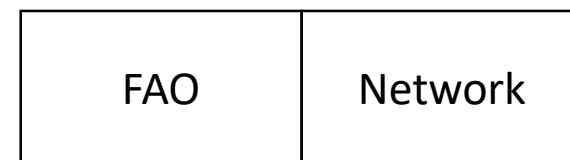
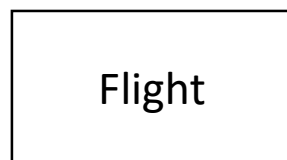
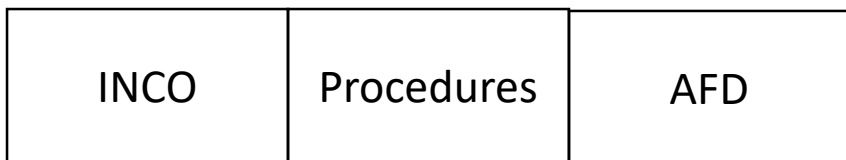
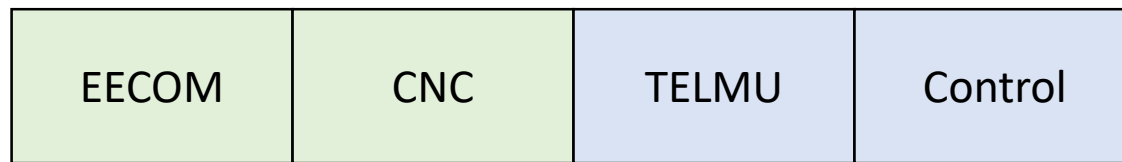
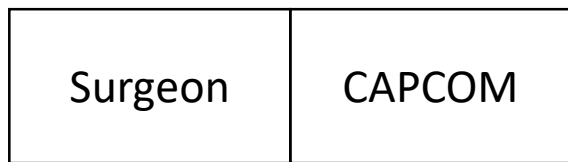
KEY

- | | | |
|-----------------------------------|-----------------------------------|--|
| ① Booster Systems Engineer | ⑪ Ass't Flight Director | ⑳ Telephone/Press Booths |
| ② Retrofire Officer | ⑫ Flight Director | ㉑ Mission Control Main Floor |
| ③ Flight Dynamics | ⑬ Experimenting & Flight Planning | ㉒ Display Screen |
| ④ Guidance Officer | ⑭ Network Controller | ㉓ Projector Plotter Display Equipment Room |
| ⑤ Flight Surgeon | ⑮ Public Affairs Officer | |
| ⑥ Spacecraft Communicator | ⑯ Flight Operations Director | |
| ⑦ Electrical | ⑰ Mission Director | |
| ⑧ Mechanical | | |
| ⑨ Life Support | | |
| ⑩ Operations & Procedures Manager | | |





“The Trench”





NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

MSC-01807
11/1/70

FINAL FLIGHT MISSION RULES

APOLLO 14
(AS-509/110/LM-8)

NOVEMBER 1, 1970

PREPARED BY
FLIGHT CONTROL DIVISION

MANNED SPACECRAFT CENTER
HOUSTON, TEXAS

FOR NASA/DOD INTERNAL USE ONLY
INCLUDING APPROPRIATE CONTRACTORS

INDEXING DATA	DATE	OPR	#	T	PGM	SUBJECT	SIGNATOR	LOC
	11-01-70	MSC	MSC-01807	R	AS-509	(206)	MSC	079-47

NASA - Manned Spacecraft Center

MISSION RULES

SECTION 3 MISSION RULE SUMMARY

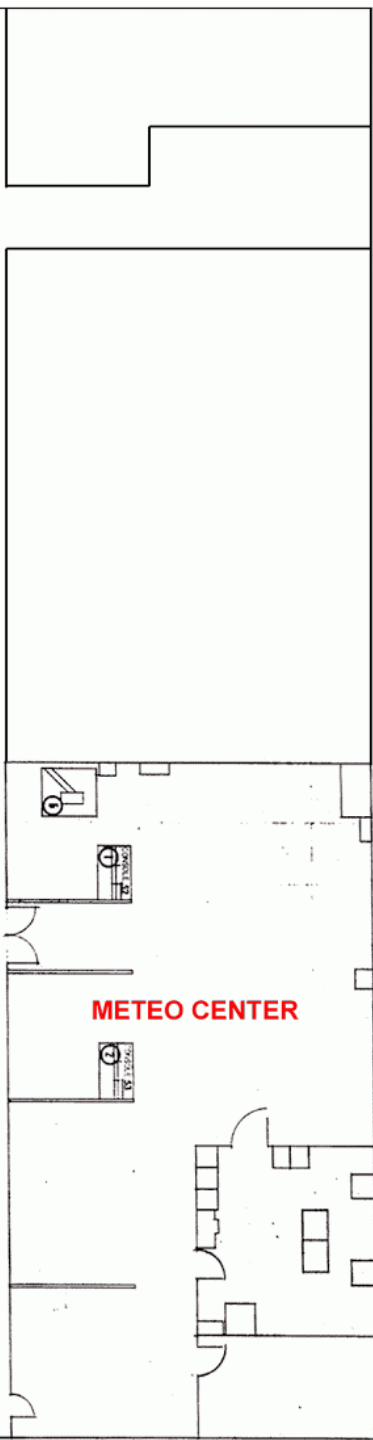
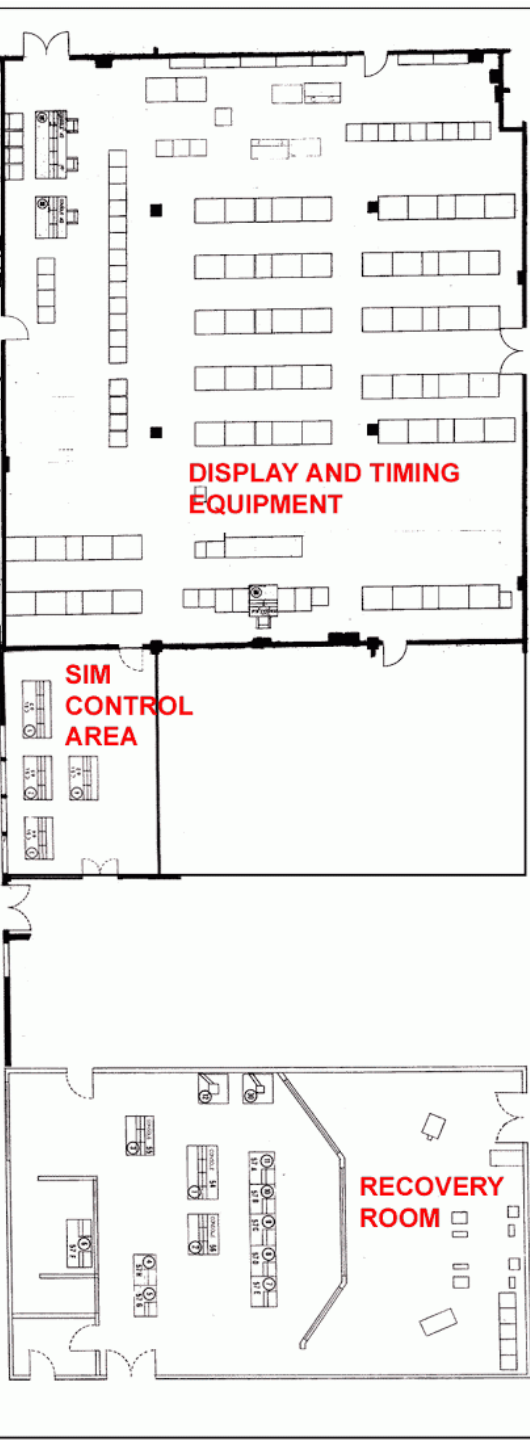
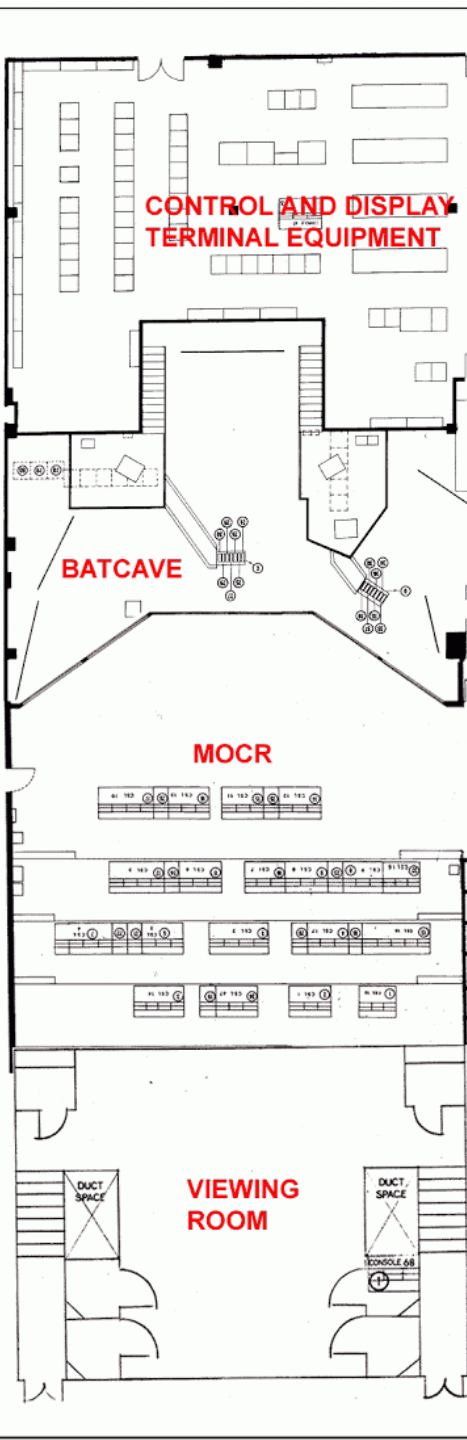
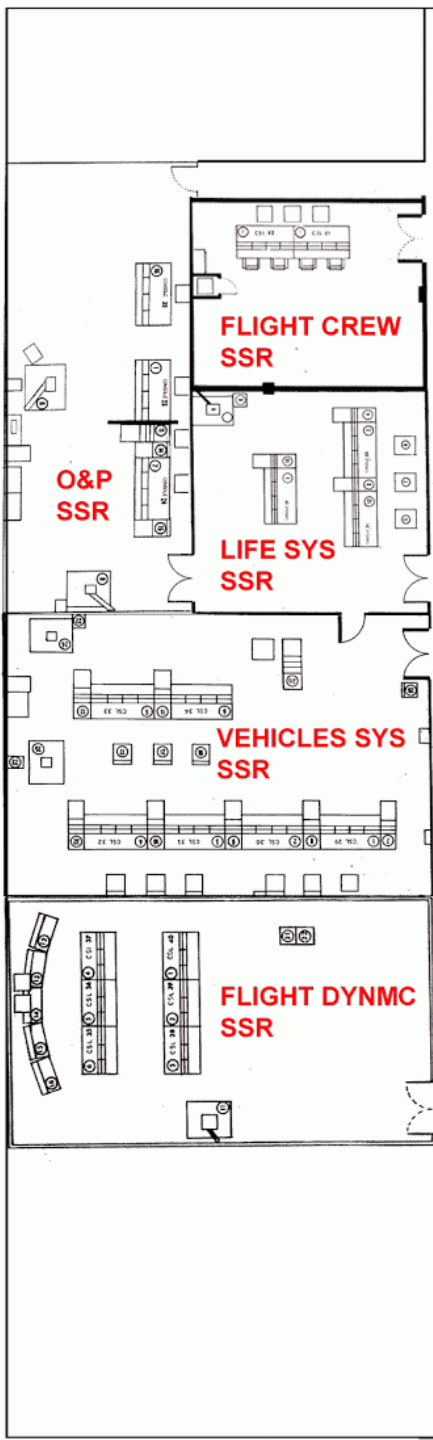
R	ITEM					
		----- ! POWERED DESCENT PHASE ! -----				
3-49	PDI IGNITION	THE FOLLOWING ACTION WILL BE TAKEN---				
	1. AUTO ULLAGE GOOD	- IF NO AUTO DPS IGN, FLIGHT CREW PERFORM MANUAL DPS IGNITION				
	2. NO AUTO ULLAGE	-FLIGHT CREW BACK UP THE ULLAGE MANEUVER -IF NO AUTO DPS IGN FLIGHT CREW WILL NO-GO PDI				
3-50	PDI TO LO GATE	POWERED DESCENT WILL BE ABORTED FOR THE FOLLOWING---				
	A. LR DATA IS REQUIRED FOR LANDING - NO LR DATA BY 10 K FT - ABORT.					
	1. LR CONVERGENCE (ALTITUDE ONLY) - DATA NOT BEING ACCEPTED OR CONVERGING FOLLOWING LOCKON FOR 60 SECONDS - ABORT.					
	2. LR DATA ACCEPTED AND CONVERGED CONTINUOUS TO P-64 - CONTINUE MISSION IF LOSS OF LOCK OCCURS IN P-64.					
	3. LR DATA ACCEPTED AND CONVERGED WITH SUBSEQUENT DROPOUT - CONTINUE TO P-64.	(A) LANDING RADAR REGAINED IN P-64. (1) DELTA H LESS THAN 1000 FT BETWEEN PGNS AND LR - CONTINUE MISSION. (2) DELTA H GREATER THAN 1000 FT BETWEEN PGNS AND LR - ATTEMPT MANUAL LANDING IN AGS. (B) LR NOT REGAINED IN P-64 - ABORT.				
	4. LATE LR LOCKON WITH DATA BEING INCORPORATED AND CONVERGING - CONTINUE TO P-64.	(A) DELTA H LESS THAN 1000 FT BETWEEN PGNS AND LR - CONTINUE MISSION. (B) DELTA H GREATER THAN 1000 FT BETWEEN PGNS AND LR - ATTEMPT MANUAL LANDING IN AGS.				
	B. PGNS ALTITUDE LESS THAN 22,000 FEET AND PGNS NAVIGATION ERRORS, CONFIRMED BY MSFN OR DOPPLER RESIDUALS, THAT CAUSE THE AGS-PGNS RADIAL VELOCITY DIFFERENCE TO EXCEED MINUS 10 FPS, PRIOR TO LANDING RADAR ALTITUDE INCORPORATION AND CONVERGENCE (A MINUS VELOCITY DIFFERENCE INDICATES THAT THE AGS TRAJECTORY IS LOWER THAN THE PGNS TRAJECTORY).					
	C. PGNS NAVIGATION ERRORS, CONFIRMED BY MSFN OR DOPPLER RESIDUALS, THAT RESULT IN THE FOLLOWING AGS-PGNS VELOCITY DIFFERENCES---	DELTA X DOT (DOWNRANGE) GREATER THAN +/- 45 FPS DELTA Y DOT (CROSSRANGE) GREATER THAN +/- 90 FPS DELTA Z DOT (RADIAL) GREATER THAN +/- 35 FPS				
	D. PGNS ALTITUDE LESS THAN 18,000 FEET AND PGNS NAVIGATION ERRORS, CONFIRMED BY DOPPLER BUT NOT BY AGS, CAUSE THE MSFN-PGNS RADIAL VELOCITY DIFFERENCE TO EXCEED MINUS 20 FPS PRIOR TO LANDING RADAR ALTITUDE INCORPORATION AND CONVERGENCE.					
	E. PGNS NAVIGATION ERRORS CONFIRMED BY DOPPLER RESIDUALS BUT NOT BY AGS, THAT RESULT IN THE FOLLOWING MSFN-PGNS VELOCITY DIFFERENCES---	DELTA Y DOT (CROSSRANGE) GREATER THAN +/- 200 FPS. DELTA Z DOT (RADIAL) GREATER THAN +/- 35 FPS.				
	NOTE---RULES C AND E ARE INDEPENDENT OF ANY TYPE OF LANDING RADAR UPDATE. FOR RULES B AND C, SWITCHOVER TO AGS WILL BE PERFORMED.					
	MISSION	REV	DATE	SECTION	GROUP	PAGE
	APOLLO 14	FNL	11/1/70	MISSION RULE SUMMARY	POWERED DESCENT	3-9

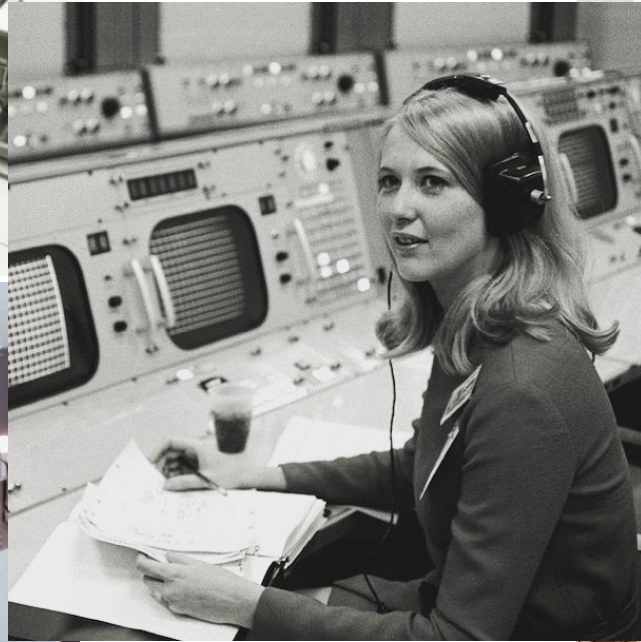
NASA - Manned Spacecraft Center

MISSION RULES

SECTION 5 TRAJECTORY AND GUIDANCE

R	ITEM					
5-89	LR DATA IS REQUIRED FOR LANDING--NO LR DATA BY 10K FT --ABORT.					
	A. LR CONVERGENCE (ALTITUDE ONLY) - DATA NOT BEING ACCEPTED OR CONVERGING FOLLOWING LOCKON FOR 60 SECONDS - ABORT.					
	B. LR DATA ACCEPTED AND CONVERGED CONTINUOUS TO P-64 - CONTINUE MISSION IF LOSS OF LOCK OCCURS IN P-64.					
	C. LR DATA ACCEPTED AND CONVERGED WITH SUBSEQUENT DROPOUT - CONTINUE TO P-64.	1. LANDING RADAR REGAINED IN P-64. (A) DELTA H LESS THAN 1000FT BETWEEN PGNS AND LR - CONTINUE MISSION. (B) DELTA H GREATER THAN 1000 FT BETWEEN PGNS AND LR - ATTEMPT MANUAL LANDING IN AGS.				
	D. LATE LR LOCKON WITH DATA BEING INCORPORATED AND CONVERGING - CONTINUE TO P-64.	1. DELTA H LESS THAN 1000 FT BETWEEN PGNS AND LR - CONTINUE MISSION. 2. DELTA H GREATER THAN 1000 FT BETWEEN PGNS AND LR - ATTEMPT MANUAL LANDING IN AGS.				
5-90	A. POWERED DESCENT WILL BE TERMINATED FOR---					
	1. PGNS ALTITUDE LESS THAN 22,000 FEET AND PGNS NAVIGATION ERRORS, CONFIRMED BY MSFN OR DOPPLER RESIDUALS, THAT CAUSE THE AGS-PGNS RADIAL VELOCITY DIFFERENCE TO EXCEED MINUS 10 FPS, PRIOR TO LANDING RADAR ALTITUDE INCORPORATION AND CONVERGENCE (A MINUS VELOCITY DIFFERENCE INDICATES THAT THE AGS TRAJECTORY IS LOWER THAN THE PGNS TRAJECTORY).					
	2. PGNS NAVIGATION ERRORS, CONFIRMED BY MSFN OR DOPPLER RESIDUALS, THAT RESULT IN THE FOLLOWING AGS-PGNS VELOCITY DIFFERENCES---	DELTA X DOT (DOWNRANGE) GREATER THAN +/- 45 FPS DELTA Y DOT (CROSSRANGE) GREATER THAN +/- 90 FPS DELTA Z DOT (RADIAL) GREATER THAN +/- 35 FPS				
	3. PGNS ALTITUDE LESS THAN 18,000 FEET AND PGNS NAVIGATION ERRORS, CONFIRMED BY DOPPLER BUT NOT BY AGS, CAUSE THE MSFN-PGNS RADIAL VELOCITY DIFFERENCE TO EXCEED MINUS 20 FPS PRIOR TO LANDING RADAR ALTITUDE INCORPORATION AND CONVERGENCE.					
	4. PGNS NAVIGATION ERRORS CONFIRMED BY DOPPLER RESIDUALS BUT NOT BY AGS, THAT RESULT IN THE FOLLOWING MSFN-PGNS VELOCITY DIFFERENCES---	DELTA Y DOT (CROSSRANGE) GREATER THAN +/- 200 FPS. DELTA Z DOT (RADIAL) GREATER THAN +/- 35 FPS.				
	NOTE---RULES 2 AND 4 ARE INDEPENDENT OF ANY TYPE OF LANDING RADAR UPDATE. FOR RULES 1 AND 2, SWITCHOVER TO AGS WILL BE PERFORMED.					
	5. COMMANDED THRUST INCREASING PRIOR TO THROTTLE DOWN OR P63 TGO=80 SEC.					
	6. RESERVED					
	7. NO THROTTLE RECOVERY BY P63/664 PROGRAM SWITCH PLUS 15 SEC.					
	8. FAILURE TO ACHIEVE FTP BY NOMINAL TIG +31 SEC. (ABORT AT GTC DIVERGENCE).					
	9. FAILURE TO ENTER P64 WHEN TGO EQUALS 60 SECONDS.					
	10. THE FOLLOWING PGNS ALARMS---20105, 00214, 20430, 20607, 21103, 01107, 21204, 21302, 21501, 00402 (CONTINUING).					
	B. POWERED DESCENT MANEUVER WILL BE TERMINATED AND AN ABORT REQUESTED IF THE TIME BIASED DPS ABORT BOUNDARY IS VIOLATED.					
5-91	AN ABORT WILL NOT BE REQUESTED FOR A PGNS FAILURE AFTER A PGNS INDICATION THAT THE HIGH GATE TARGETING CONDITIONS HAVE BEEN ACHIEVED.					
	MISSION	REV	DATE	SECTION	GROUP	PAGE
	APOLLO 14	FNL	11/1/70	TRAJECTORY AND GUIDANCE	DESCENT	5-12





Program	Number of crewmen	Simulator time, hr (a)	Simulator time per crewman (average), hr	Total training program time, hr	Simulator portion of total training program time, percent
Mercury	7	1 330	190	4 038	33
Gemini	20	6 964	348	17 991	39
Apollo (through mission 15)	32	29 967	936	69 248	43
Total	59	38 261	--	91 277	42

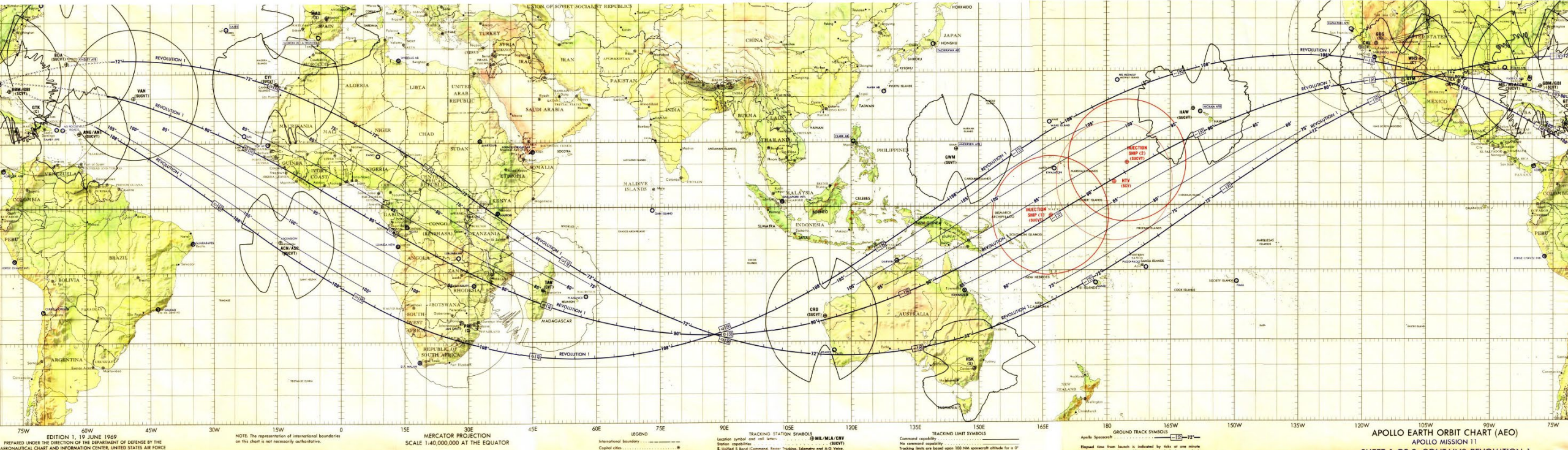
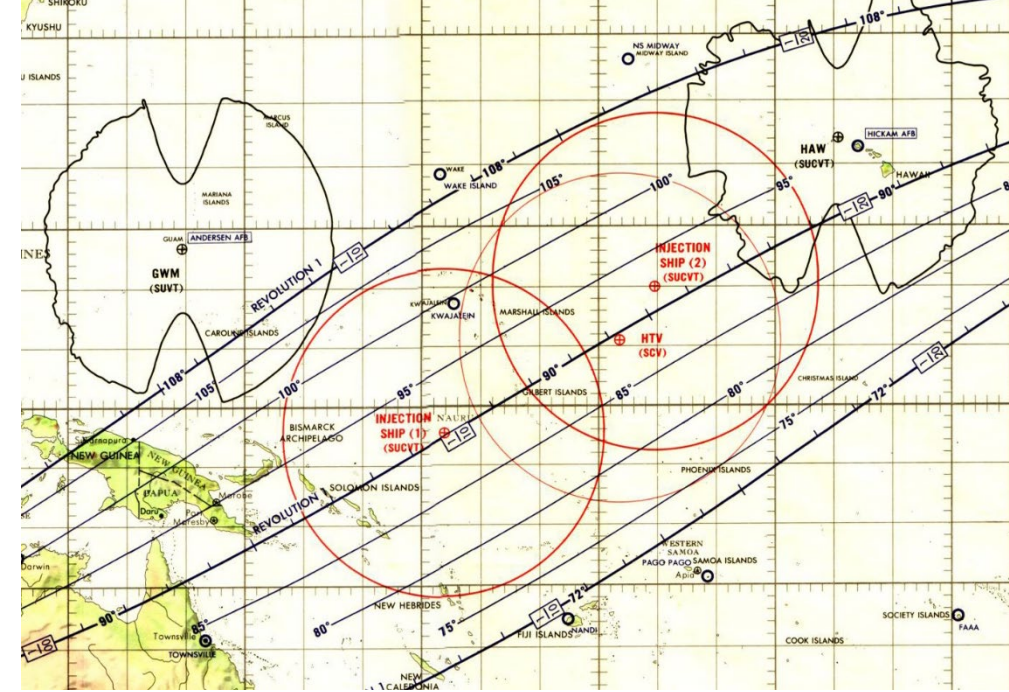
Apollo Mission	Integrated Simulation sessions, days			
	CMS/MCC	LMS/MCC	CMS/LMS/MCC	Total
7	18	0	0	18
8	14	0	0	14
9	10	2	8	20
10	11	0	7	18
11	7	4	7	18
12	10	3	12	25
13	13	5	9	27
14	15	7	13	35
15	19	5	7	31







Manned Space Flight Network



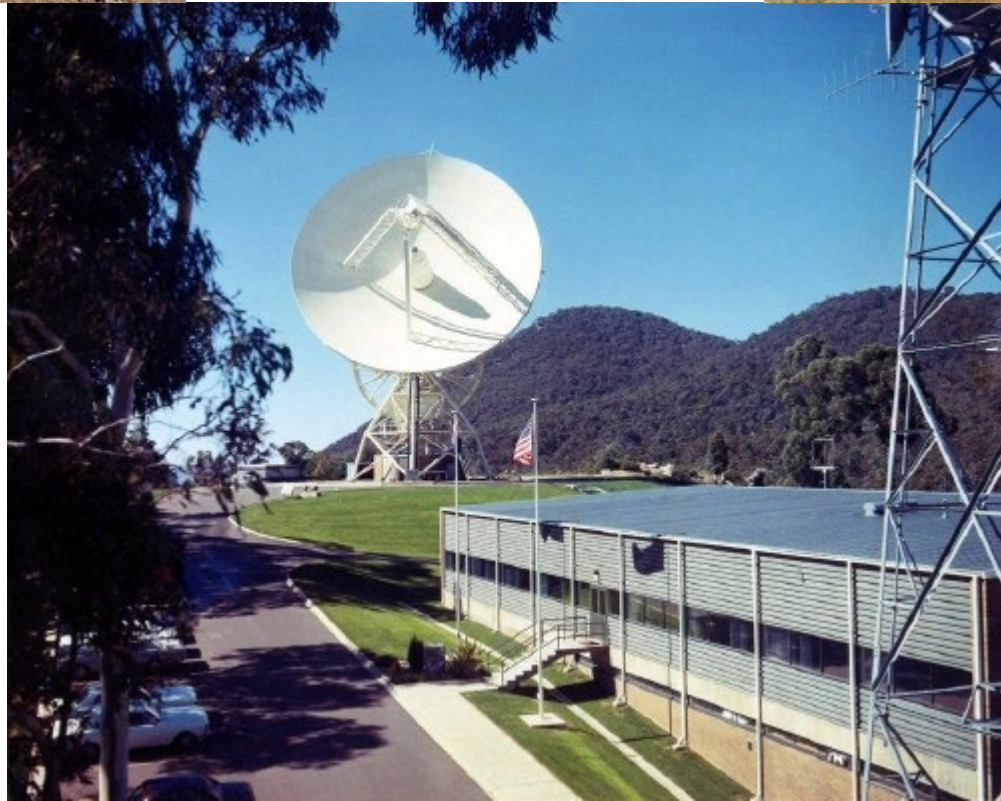
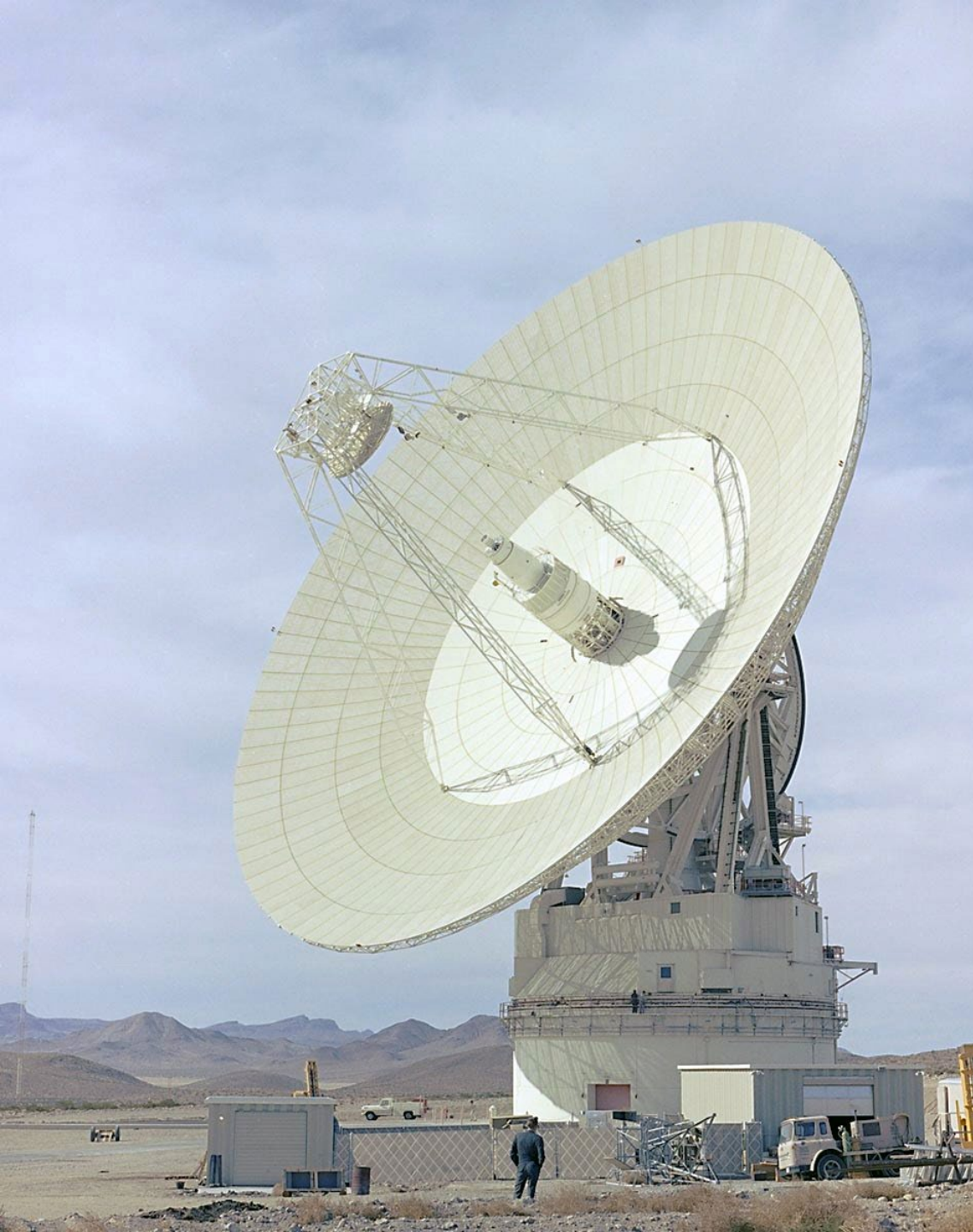
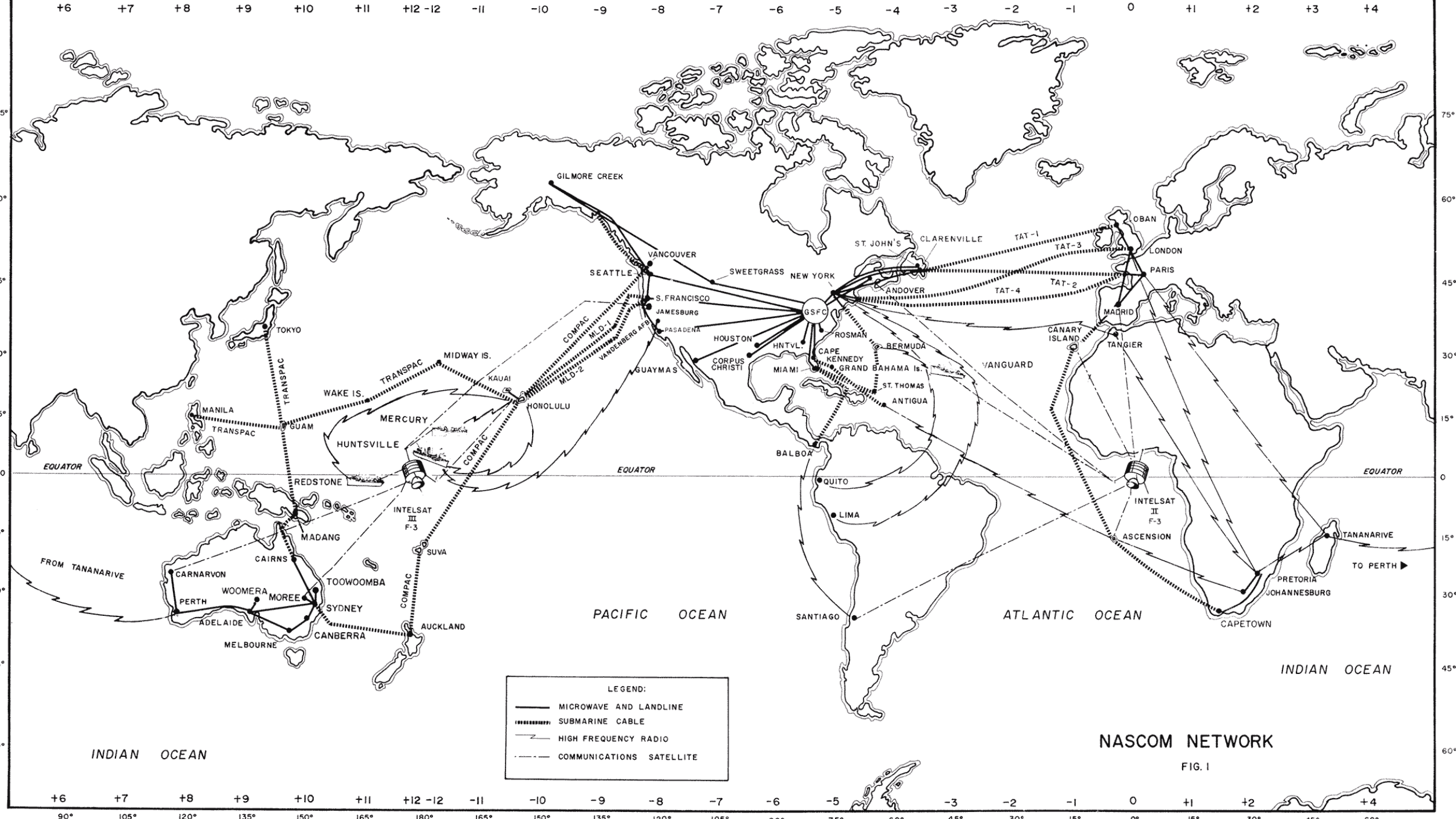


Table 1-1. Network Configuration for AS-512

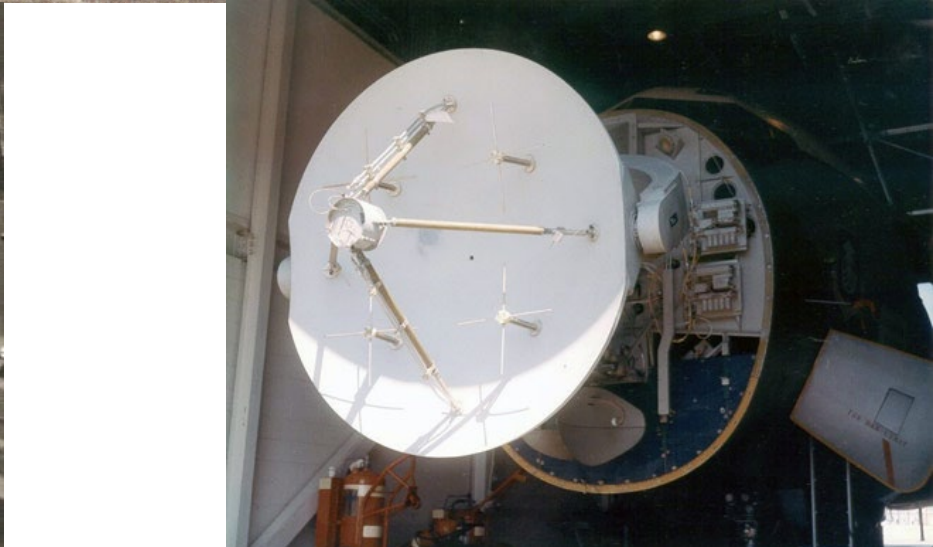
SYSTEMS STATIONS	TRACKING			USB			TLM			DATA PROCESSING			COMM			OTHER				
	C-band (High-speed)	C-band (Low-speed)	USB	TV to MCC	Voice	TLM	Command	VHF Links	Mag Tape Recording	Decoms	642B TLM	642B CMD	CDP	Acq Computer	High-speed Data	TTY	Voice (SCAMA)	Voice VHF A/G	Range Safety	SPAN
ACN			X		X	X	X	X	X	X	X		X	X	X	X	X			
ANT	X	X																X	X	
ARIA(4)					X	X		X	X							X	X	X		
AOCC																X	X			
BDA	X	X	X		X	X	X	X	X	X	X		X	X	X	X	X	X	X	
CNV	X	X																	X	
CRO	X	X	X		X	X	X	X	X	X	X		X	X	X	X	X	X		X
CYI			X		X	X	X	X	X	X	X		X	X	X	X	X	X		X
GBI	X	X																	X	
GDS			X	X	X	X	X		X	X	X		X	X	X	X	X	X		
PIR			X		X	X	X						X							
GTK	X	X																	X	
GWM			X		X	X	X	X	X	X	X		X	X	X	X	X	X		
HAW			X		X	X	X	X	X	X	X		X	X	X	X	X	X		
HSK			X	X	X	X	X		X	X	X		X	X	X	X				
NBE			X		X	X	X						X							
MAD			X	X	X	X	X		X	X	X		X	X	X	X				
RID			X		X	X	X						X							
MARS				X	X	X														
MIL			X		X	X	X	X	X	X	X		X	X	X	X	X	X		
MLA	X	X																	X	
PARKES				X	X	X														
TEX			X		X	X	X	X	X	X	X		X	X	X	X	X	X		
VAN	X	X	X		X	X	X	X	X	X	X	X	X	X	X	X	X	X		
ETC			X		X	X	X		X	X	X					X	X			



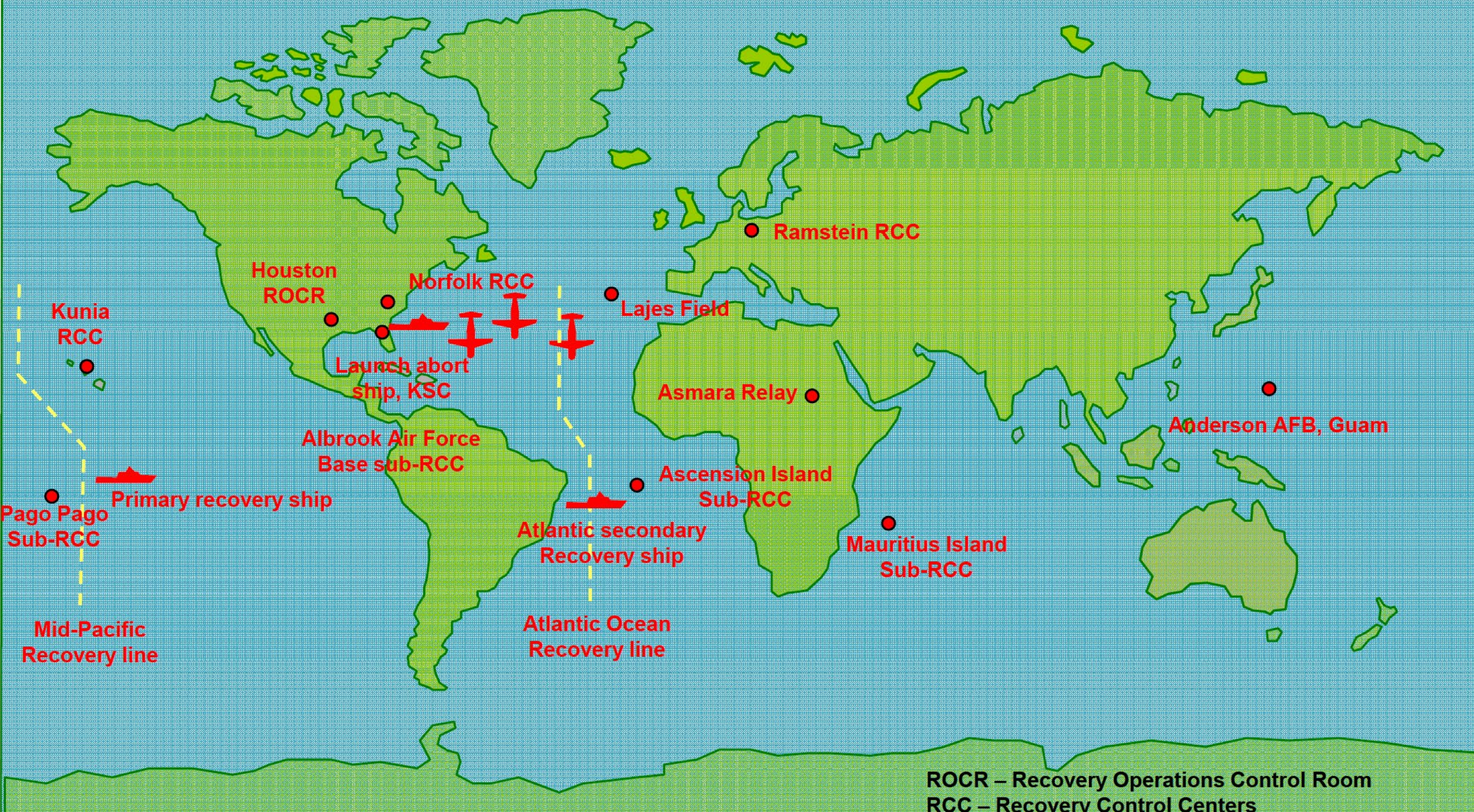


NASCOM NETWORK

FIG. 1







ROCR – Recovery Operations Control Room
RCC – Recovery Control Centers

Discussion Groups

- Cox Chapters 18 and 19 (“We’re Going to Put a Guy in That Thing and Light It”, “There Will Always Be People Who Want to Work in That Room”)
 - The creation of Mission Control
- Mission Control video
 - Interviews with several flight controllers
- Woodling (“Simulation of Manned Space Flight for Crew Training”)
 - The development and use of spacecraft simulators
- Tsiao Chapter 5 (“The Apollo Years”)
 - Creation of the Manned Space Flight Network

BOOSTER	RETRO
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FIDO	GUIDO
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SURGEON	CAPCOM
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EECOM	CNC	TELMU	CONTROL
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INCO	O&P	AFD
------	-----	-----

FLIGHT

FAO	NETWORK
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PAO

FOD

MD

DoD

BOOSTER SYSTEMS
ENGINEER



LOC	DESCRIPTION	TYPE	NOTE	LOC	DESCRIPTION	TYPE
01	EVENT INDICATOR	D9/5B		14	STATUS/STATUS REPORT	D9/1A
02	EVENT INDICATOR	D9/5B		15	TOGGLE SWITCH/INDICATOR	D9/9A
03	EVENT INDICATOR	D9/5B		16	MANUAL SELECT KEYBOARD	A6B/5
04	7 DIGIT CLOCK	D8/3		17	SUMMARY MSG ENABLE KEYBOARD	A19/A
05	EVENT INDICATOR	D9/5B		18	VOICE COMM POSITION-3018	H48MFD
07	VOICE COMM POSITION-3016	V48MFD		19	MANUAL SELECT KEYBOARD	A6B/6
08	TV MONITOR 14" PRECISION	C2/1		21	SWITCH MODULE	D9/40F
09	TV MONITOR 14" PRECISION	C2/1		23	SWITCH MODULE	D9/40E
10	TV MONITOR 14" PRECISION	C2/1		24	SWITCH MODULE	D9/40F
11	VOICE COMM POSITION-3017	V48MFD		25	LOAD NUMBER INDICATOR	D9/41B
12	EVENT INDICATOR (72)	D9/28				

